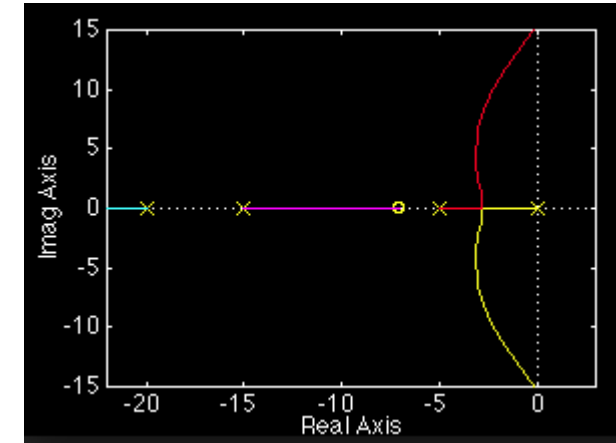
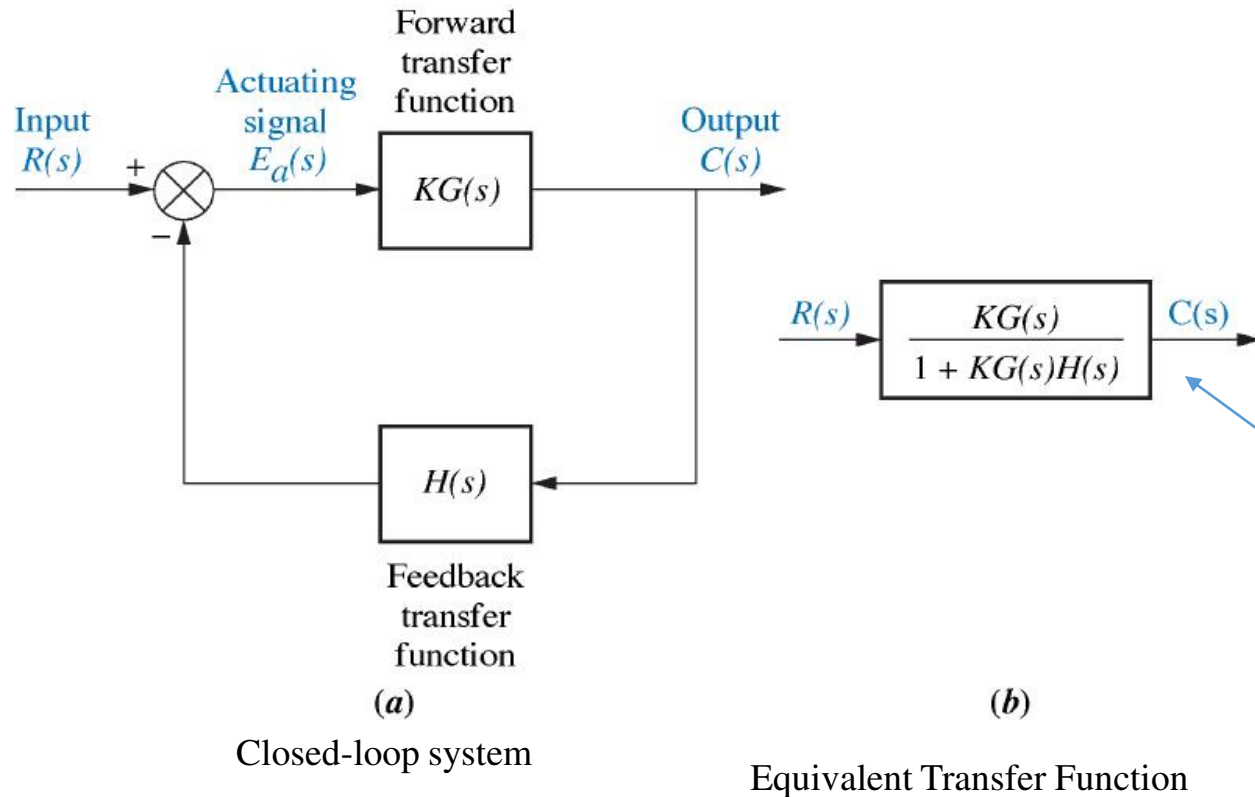


Design via Root Locus

1. Root Locus.
2. Compensator design via Root Locus.
3. Physical Realization of Compensation.

Root Locus Techniques

- Root locus is a graphical presentation of the closed-loop poles as a system parameter k is varied.
- The graph of all possible roots of this equation (K is the variable parameter) is called the root locus.
- The root locus gives information about the stability and transient response of feedback control systems.



Root-locus (poles motion graph)

Zeros of $T(s)$ are zeros of $G(s)$ and poles of $H(s)$.

Poles of $T(s)$ depends on gain K

CLCF is a function of K .

Root Locus graphically shows poles of $T(s)$ as K varies

Evaluation of a Complex Function via Vectors

Any complex number, $\sigma + j\omega$, described in Cartesian coordinates can be graphically represented by a vector,

Magnitude and phase Of $F(s)$ at S

If $F(s) = \frac{\prod_{i=1}^m (s+z_i)}{\prod_{j=1}^n (s+p_j)}$ \rightarrow
$$\left\{ \begin{aligned} M &= \frac{\prod \text{zero lengths}}{\prod \text{pole lengths}} = \frac{\prod_{i=1}^m |(s+z_i)|}{\prod_{j=1}^n |(s+p_j)|} \\ \theta &= \sum \text{zero angles} - \sum \text{pole angles} = \sum_{i=1}^m \angle (s+z_i) - \sum_{j=1}^n \angle (s+p_j) \end{aligned} \right.$$

Problem: Given $F(s) = \frac{(s+1)}{s(s+2)}$ Find $F(s)$ at the point $s = -3 + j4$

Solution: Any complex number can be represented by a vector

For zero (point $s_1 = -1$) the vector is:

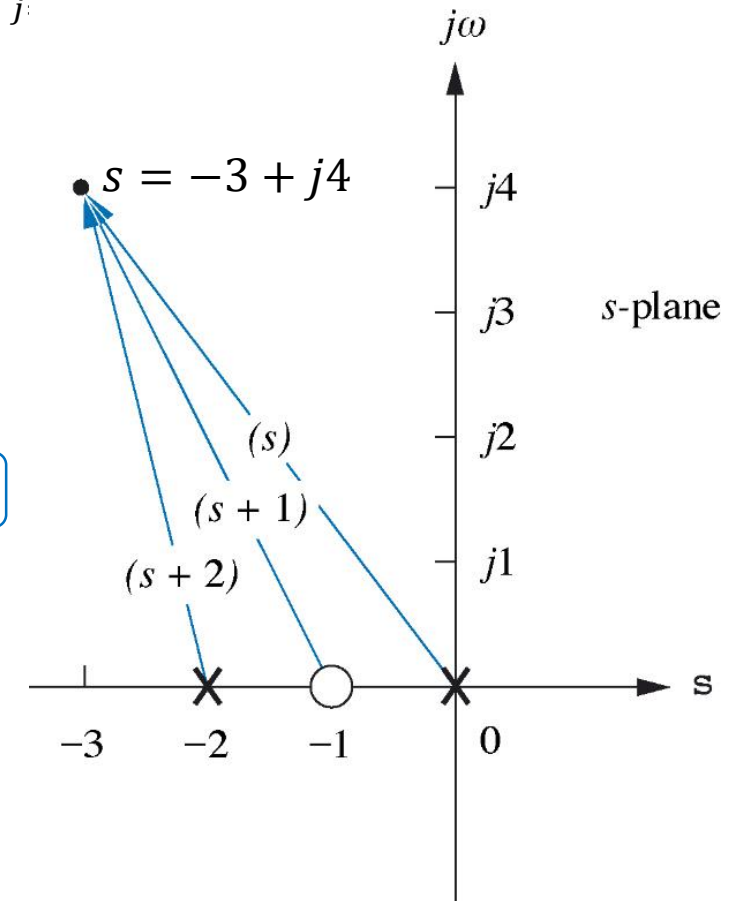
$$s - s_1 = s - (-1) = (-3 + j4) - (-1) = -2 + j4 = \sqrt{(-2)^2 + (4)^2} \tan^{-1} \left(\frac{4}{-2} \right) = \sqrt{20} \angle 116.6^\circ$$

For pole at 0: $5 \angle 126.9^\circ$ For pole at -2: $\sqrt{17} \angle 104.0^\circ$

Vector magnitude, $M = \frac{\sqrt{20}}{5\sqrt{17}} = 0.217$

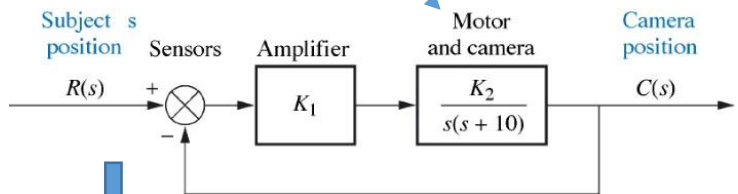
Vector angle, $\theta = \angle(116.6^\circ - 126.9^\circ - 104.0^\circ)$

Vector $M \angle \theta = 0.217 \angle -114.3^\circ$

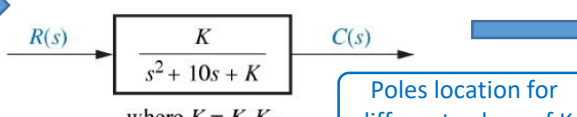




Defining the Root Locus



(b)



(c)

Poles location for different values of K

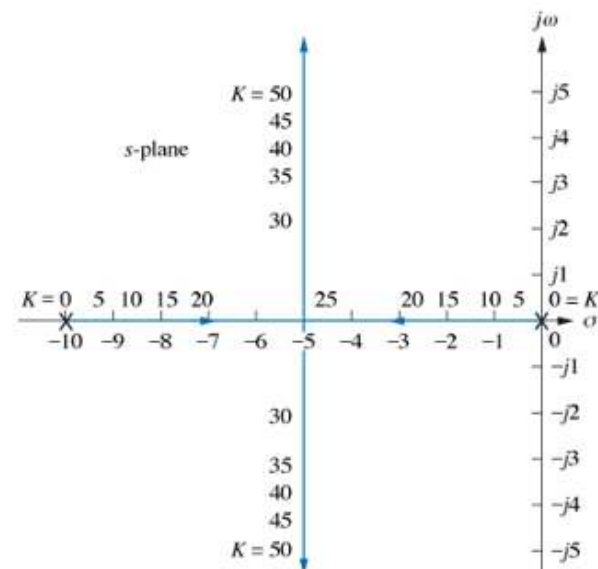
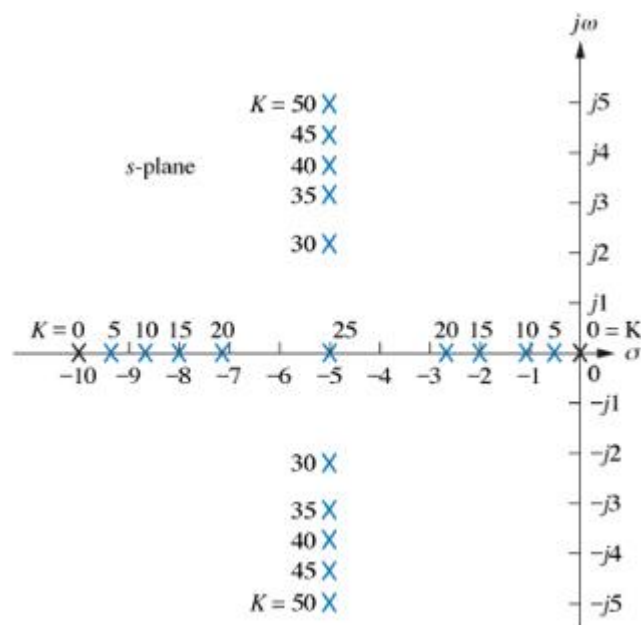
K	Pole 1	Pole 2
0	-10	0
5	-9.47	-0.53
10	-8.87	-1.13
15	-8.16	-1.84
20	-7.24	-2.76
25	-5	-5
30	-5 + j2.24	-5 - j2.24
35	-5 + j3.16	-5 - j3.16
40	-5 + j3.87	-5 - j3.87
45	-5 + j4.47	-5 - j4.47
50	-5 + j5	-5 - j5

Gain less than 25, over-damped.

Gain = 25, critically damped.

Gain over 25, under-damped.

Stable system, as no pole on right-hand plane.



During underdamped, real parts are same; so settling time (which is related to real part) remains the same.

Damping frequency (imaginary part) increases with gain, resulting in reduction of peak time.

Properties of Root Locus

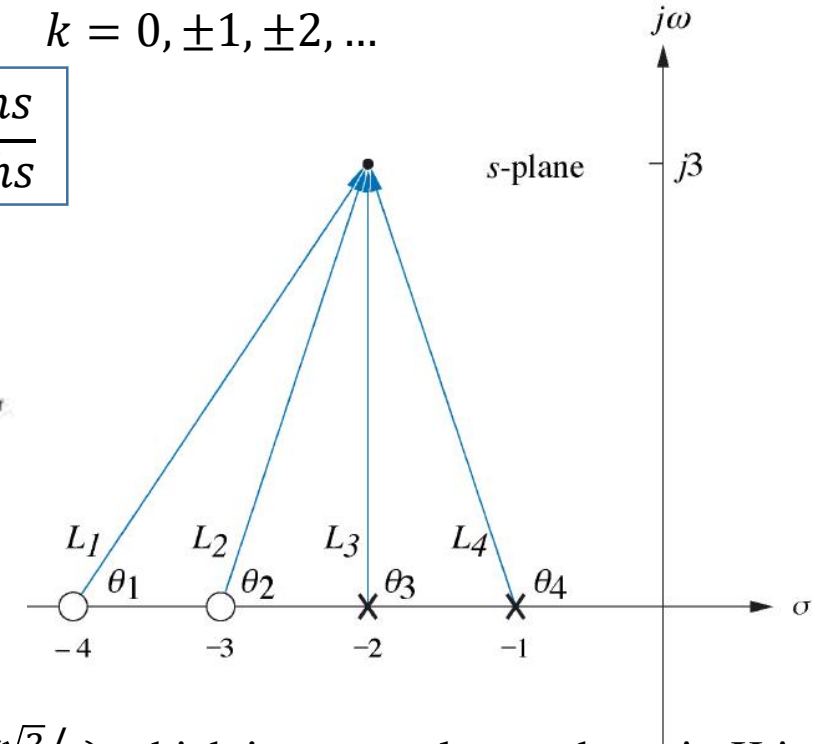
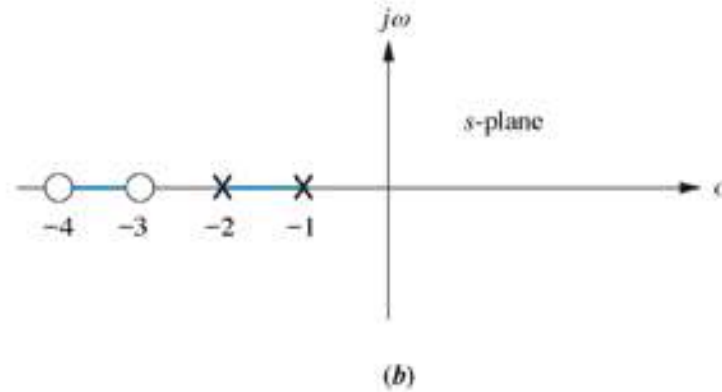
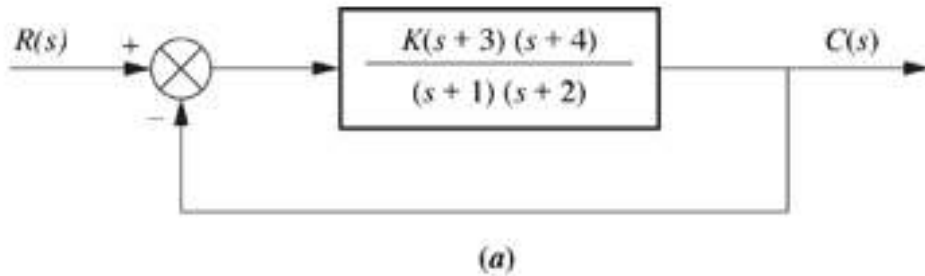
The closed-loop transfer function $T(s) = \frac{KG(s)}{1+KG(s)H(s)}$

Module condition

Angle condition

s_0 is a pole if $1 + KG(s_0)H(s_0) = 0 \Rightarrow KG(s_0)H(s_0) = -1 = 1 \angle (2k+1)180^\circ \quad k = 0, \pm 1, \pm 2, \dots$

$$KG(s)H(s) = -1 \Rightarrow |KG(s)H(s)| = 1 \Rightarrow K = \frac{1}{|G(s)||H(s)|} = \frac{\prod \text{pole lengths}}{\prod \text{zero lengths}}$$



Find if the point $-2+j3$ is on root locus for some value of gain, K :

From the angle condition

Σ zero angle - Σ pole angle

$$\theta_1 + \theta_2 - \theta_3 - \theta_4 = 56.31^\circ + 71.57^\circ - 90^\circ - 108.43^\circ = -70.55^\circ$$

Not a multiple of 180° . So, $-2 + j3$ is not in the root locus (can not be a pole for some value of K).

For the point $-2 + j(\sqrt{2}/2)$ which is on root locus, the gain K is:

$$K = \frac{\prod \text{pole lengths}}{\prod \text{zero lengths}} = \frac{L_3 L_4}{L_1 L_2} = \frac{\frac{\sqrt{2}}{2} (1.22)}{(2.12)(1.22)} = 0.33$$

Sketching the Root Locus₁

1. Number of branches: Equals the number of closed loop poles.

2. Symmetry: Symmetrical about the real axis (conjugate pairs of poles, real coefficients of the characteristic equation polynomial).

3. Real axis segments: For $K > 0$, root locus exists to the left of an odd number real axis poles and/or zeros (angle condition).

4. Start and end points: The root locus begins at finite and infinite poles of $G(s)H(s)$ and ends at finite and infinite zeros of $G(s)H(s)$.

5. Asymptotes: The root locus approaches straight lines as asymptotes as the locus approaches infinity. the equation of the asymptotes is given by:

$$\sigma_a = \frac{\sum \text{finite poles} - \sum \text{finite zeros}}{\neq \text{finite poles} - \neq \text{finite zeros}}$$

Intersection with Real axis

$$\theta_a = \frac{(2k + 1)\pi}{\neq \text{finite poles} - \neq \text{finite zeros}} \text{ for } k = 0, \pm 1, \pm 2, \dots$$

Angle in radian of the asymptote with real axis

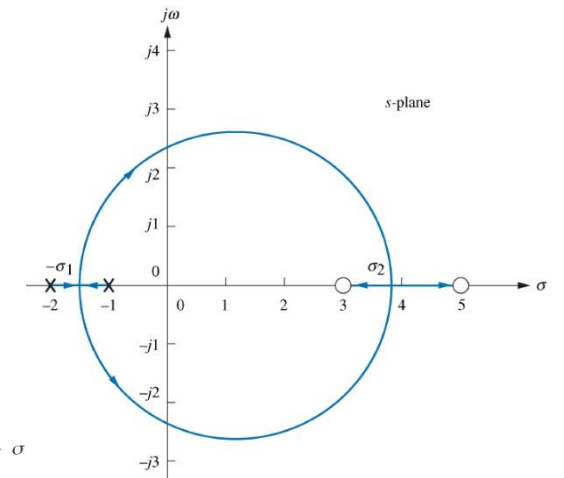
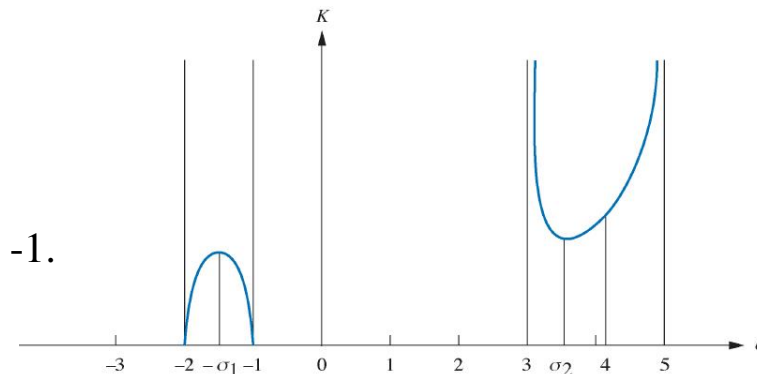
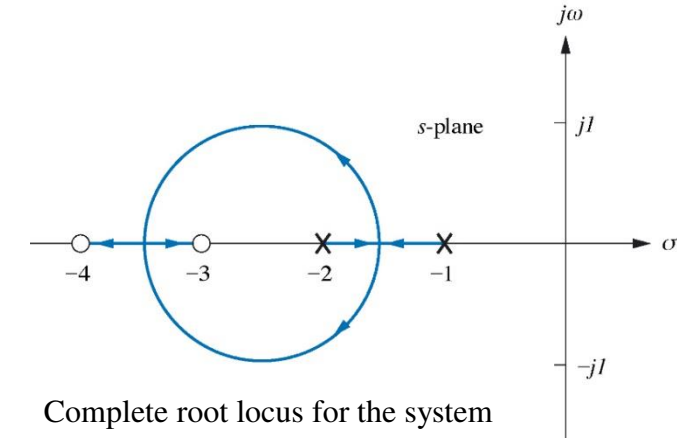
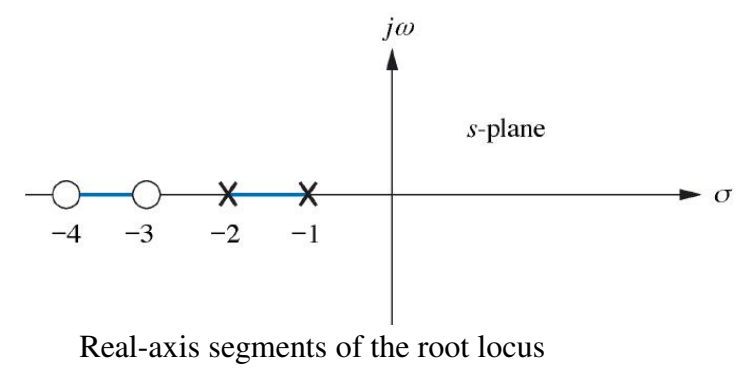
6. Real-Axis Breakaway and Break-in Points:

σ_1 : Breakaway point (leave the real axis);

σ_2 : Break-in point (return to the real axis); .

Breakaway point: at *maximum gain on the real axis* between -2 and -1.

Break-in point: at *minimum gain on real axis* (increases when moving towards a zero) between +3 and +5.



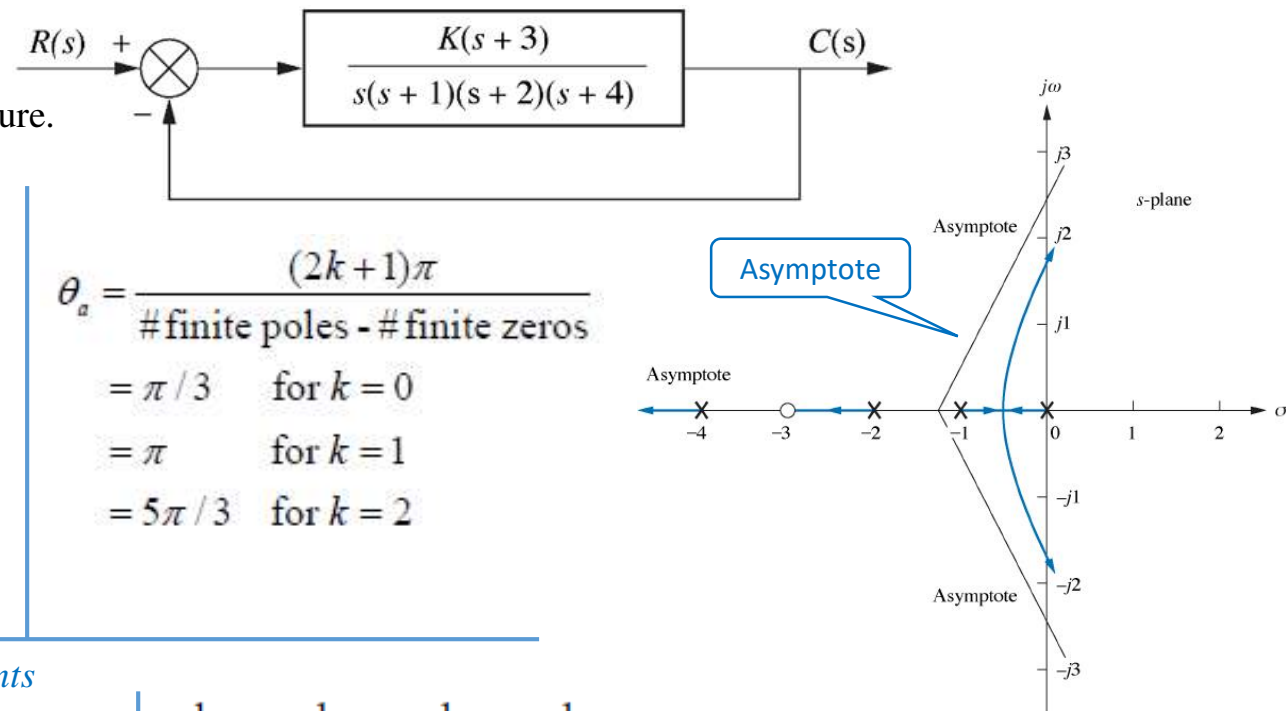
Sketching the Root Locus₂

Problem1: Sketch the Root Locus for the system shown in the following figure.

Solution1: Calculate asymptotes to find real axis intercept:

$$\sigma_a = \frac{\sum \text{finite poles} - \sum \text{finite zeros}}{\# \text{finite poles} - \# \text{finite zeros}} = \frac{(-1-2-4) - (-3)}{4-1} = -\frac{4}{3} \quad \text{and}$$

- The angles of the lines that intersect at $-4/3$ is given by θ_a :
- For higher values of k , the angles would begin to repeat.
- There are four poles and one finite zero. Root locus begins at poles and ends at zeros. (Three zeros at infinity are at the ends of the asymptotes.)



Problem2: From root-locus graph on figure find break-in and break-away points

Method 1: (transition method)

$$\sum_{i=1}^m \frac{1}{\sigma + z_i} = \sum_{i=1}^n \frac{1}{\sigma + p_i} \quad \text{where } z_i \text{ and } p_i \text{ are negative of zero and pole values, respectively, of } G(s)H(s).$$

From figure, we get

$$\begin{aligned} \frac{1}{\sigma-3} + \frac{1}{\sigma-5} &= \frac{1}{\sigma+1} + \frac{1}{\sigma+2} \\ \Rightarrow 11\sigma^2 - 26\sigma - 61 &= 0 \\ \Rightarrow \sigma &= -1.45, 3.82 \end{aligned}$$

Method 2: (Differentiation method)

$$KG(s)H(s) = \frac{K(s-3)(s-5)}{(s+1)(s+2)} = \frac{K(s^2 - 8s + 15)}{(s^2 + 3s + 2)}$$

Along the real axis ($s = \sigma$) and $KG(s)H(s) = -1$

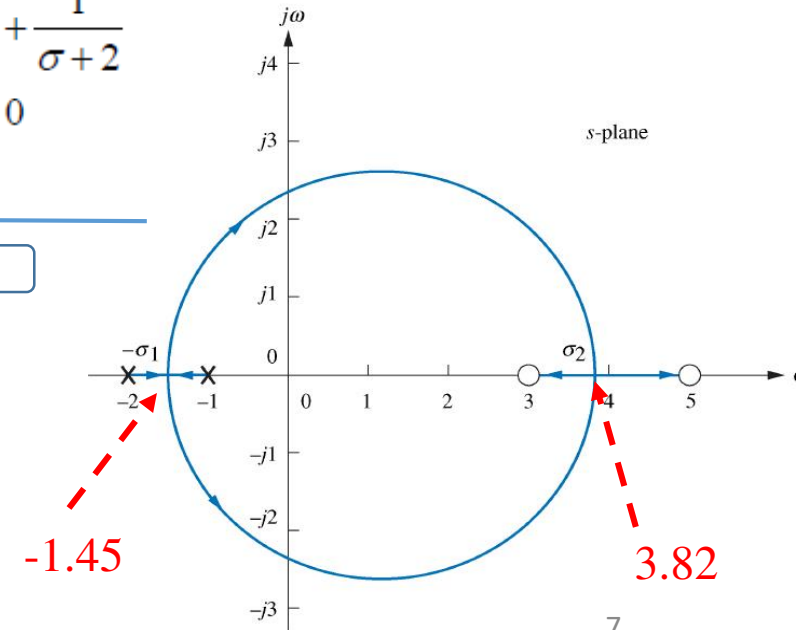
$$\frac{K(s^2 - 8s + 15)}{(s^2 + 3s + 2)} = -1 \quad \xrightarrow{\text{Solving for } K} \quad K = \frac{-(s^2 + 3s + 2)}{(s^2 - 8s + 15)}$$

Differentiating K with respect to σ (max and min)

$$\frac{dK}{d\sigma} = \frac{(11\sigma^2 - 26\sigma - 61)}{(\sigma^2 - 8\sigma + 15)^2} = 0$$

Solving for σ ,

$$\sigma = -1.45 \quad \text{and} \quad \sigma = 3.82$$



Sketching the Root Locus₃

7. Imaginary-Axis Crossing

Stability: the system's poles are in the left half-plane up to a particular value of gain K .

PROBLEM: For the system, find the frequency and gain, K , for which the root locus crosses the imaginary axis. For what range of K is the system stable?

SOLUTION: The closed-loop transfer function

$$T(s) = \frac{K(s+3)}{s^4 + 7s^3 + 14s^2 + (8+K)s + 3K}$$

Characteristic Eq.: $s(s+1)(s+2)(s+4) + K(s+3)$

$$= s^4 + 7s^3 + 14s^2 + (8+K)s + 3K = 0$$

We get Routh table as follows:

s^4	1	14	$3K$
s^3	7	$8+K$	
s^2	$(90-K)/7$	$3K$	
s^1	$\frac{-K^2 - 65K + 720}{90 - K}$		
s^0	$3K$		

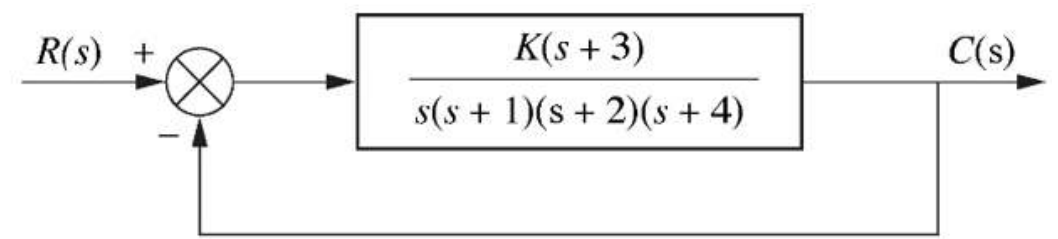
A complete *row of zeros* yields the possibility for imaginary-axis roots.

For $K > 0$, only s^1 row can be zero.

$$\begin{aligned} -K^2 - 65K + 720 &= 0 \\ \Rightarrow K &= 9.65 \end{aligned}$$

Forming the even polynomial by using the s^2 row (above) with $K = 9.65$,
 $(90 - K)s^2 + 21K = 80.35s^2 + 202.7 = 0$ Gives $s = \pm j1.59$

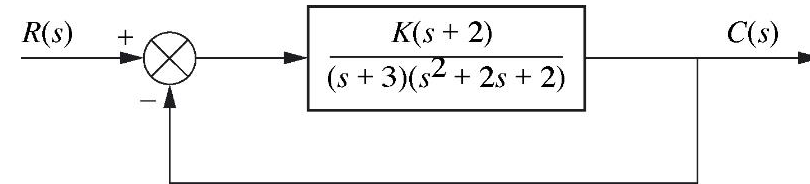
Thus, the root locus crosses the imaginary-axis at $\omega_d = \pm j1.59$ at a gain of $K = 9.65$. So, the system is stable for $0 \leq K < 9.65$



8. Angles of Departure and Arrival

Departure: from complex poles.

Arrival: to complex zeros.

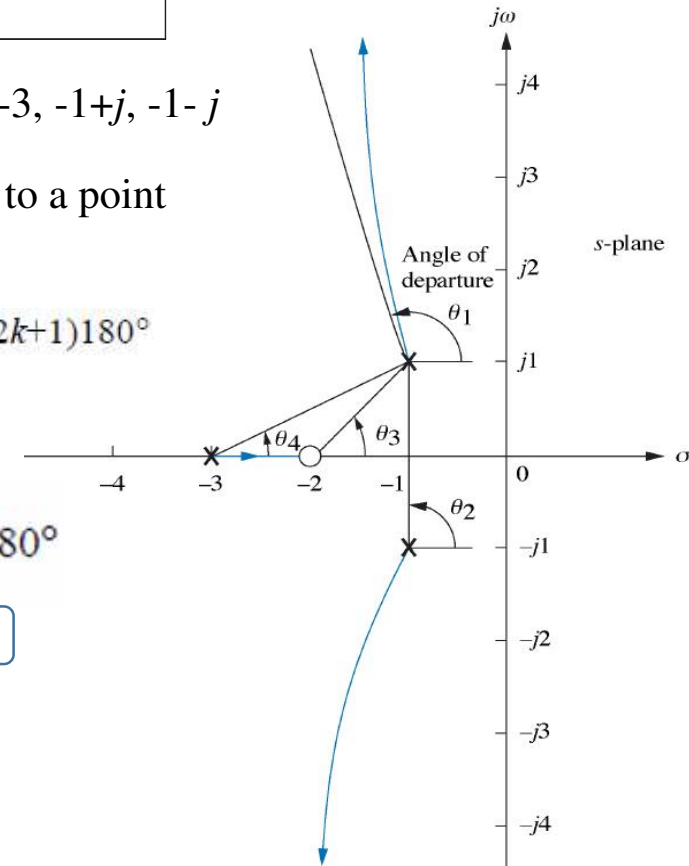


Angle of departure: Zero: -2 Poles: $-3, -1+j, -1-j$

we calculate the sum of angles drawn to a point ϵ close to the complex pole, $-1 + j$

$$\text{Sum (zero angles)} - \text{Sum (pole angles)} = (2k+1)180^\circ$$

$$\begin{aligned} -\theta_1 - \theta_2 + \theta_3 - \theta_4 &= \\ &= -\theta_1 - 90^\circ + \tan^{-1}\left(\frac{1}{1}\right) - \tan^{-1}\left(\frac{1}{2}\right) = 180^\circ \\ \Rightarrow \theta_1 &= -251.6^\circ = 108.4^\circ \end{aligned}$$



Improving System Response

Speed up the response : move pole from A to B without affecting the percent overshoot

Solution: move the root locus to put the desired pole on it *for some value of gain k* (compensation by adding poles and zeros).

- Dynamic compensator is used if a satisfactory design cannot be obtained by adjustment of gain k alone.

Compensators

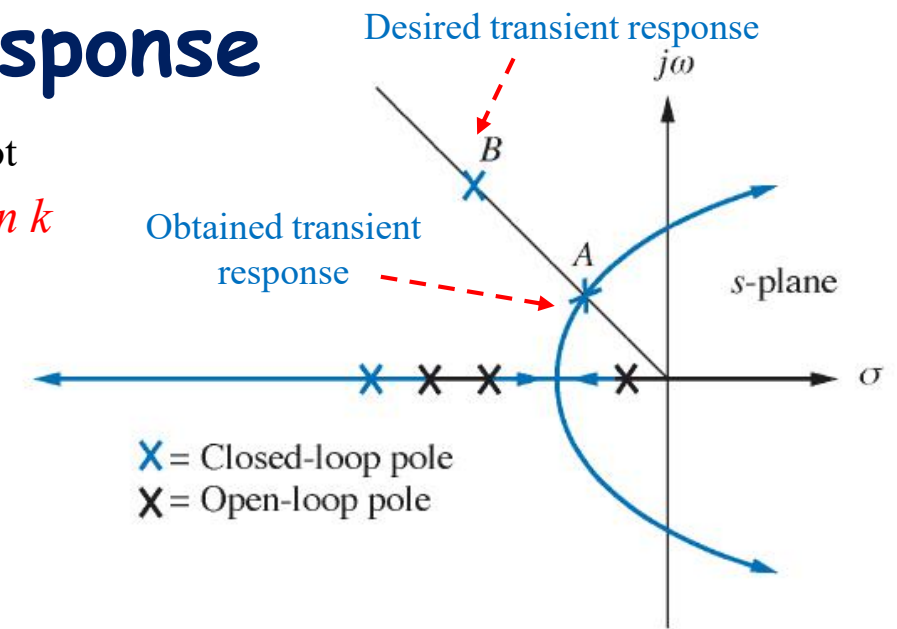
Dynamic *compensators* (function of s) *are designed to improve*:

Transient response by adding an ideal compensator **PD** (pure differentiation using active amplifiers) or a **Lead** compensator (implemented with passive elements) in the *forward* path or *feedback* path.

Steady-state error by adding an ideal compensator **PI** (pure integration using active amplifiers) or a **Lag** compensator (implemented with passive elements) in the *forward path* or *feedback path*.

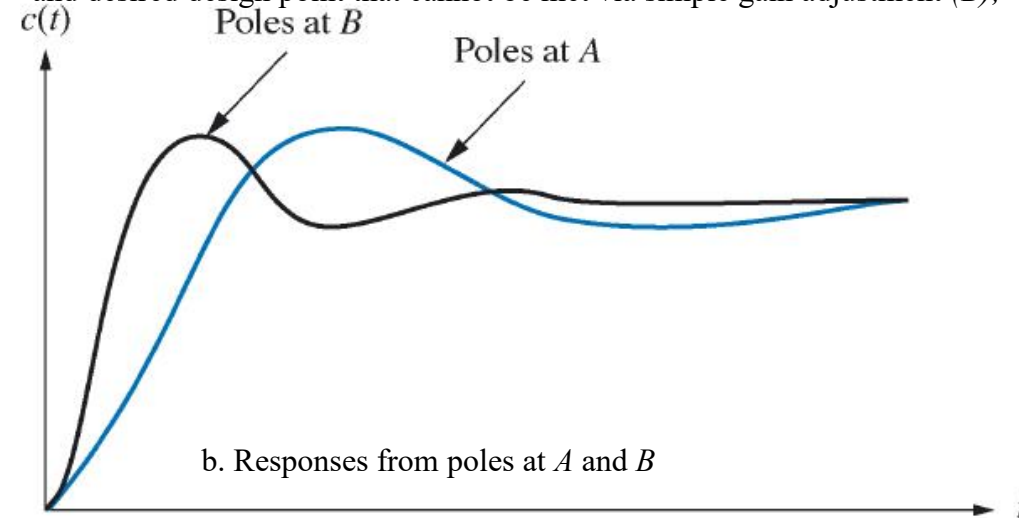
- Compensator transfer function : $C(s) = K \frac{s + z}{s + p}$

lead compensation if $z < p$ and *lag compensation* if $z > p$.



(a)

a. Sample root locus, showing possible design point via gain adjustment (A) and desired design point that cannot be met via simple gain adjustment (B);



b. Responses from poles at A and B

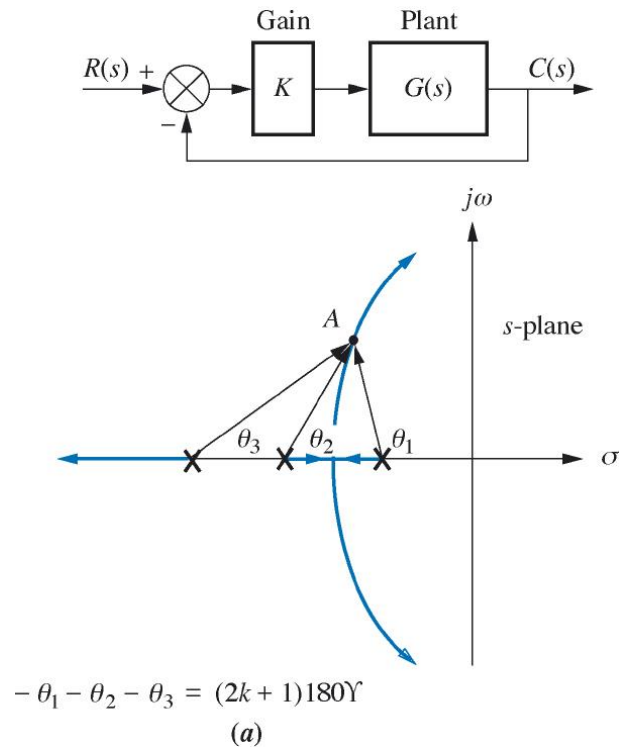
(b)

Ideal Integral Compensation (PI)

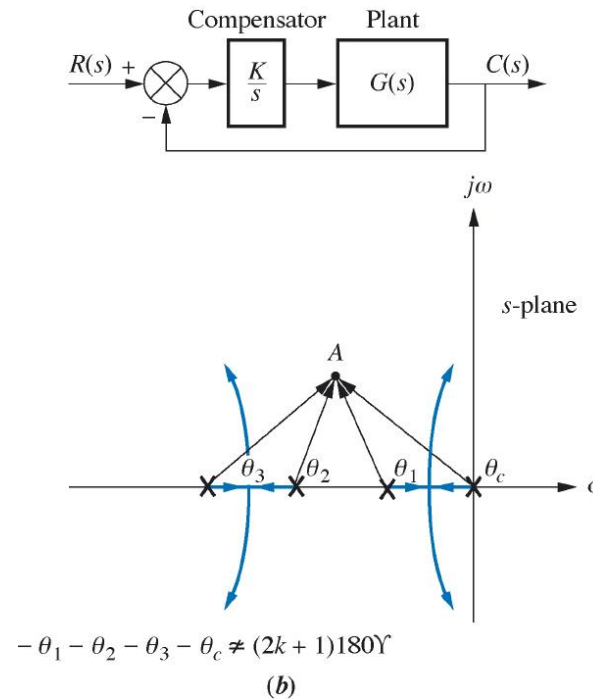
Improving Steady-State Error

- Steady-state error can be improved (without appreciably affecting the transient response) by placing an open-loop pole at the origin, because this increases the system type by one.

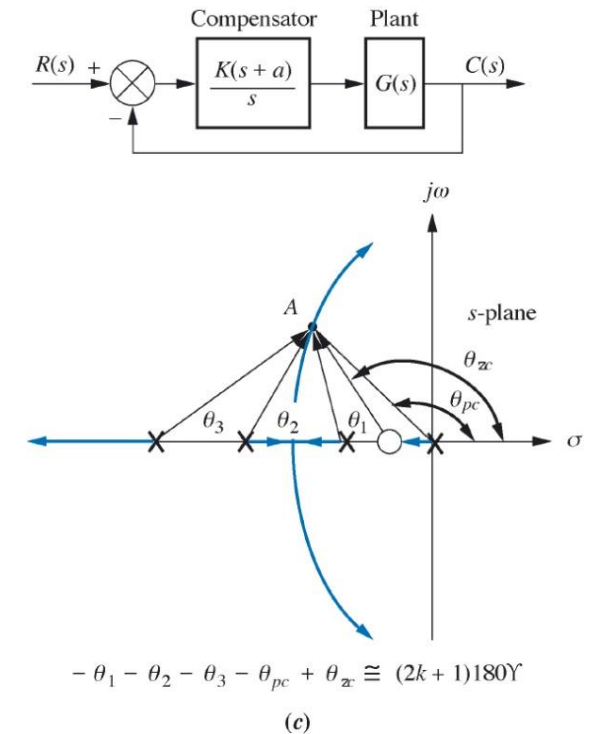
System operating with closed-loop poles at A (desirable transient response)



If we add a pole at the origin → Changes root-locus (Point A not on root locus.)



Solution: add a zero close to the pole at the origin to pole cancel out the effect of the added pole on the root-locus.



Pole at A is:

- on the root locus without compensator;
- not on the root locus with compensator pole added;
- approximately on the root locus with compensator pole and zero added

we have improved the steady-state error without appreciably affecting the transient response

Example 1

Given the system of Figure (a), operating with a *damping ratio of 0.174*, show that the addition of the ideal integral compensator shown in Figure (b) reduces the steady-state error to zero for a step input without appreciably affecting transient response.

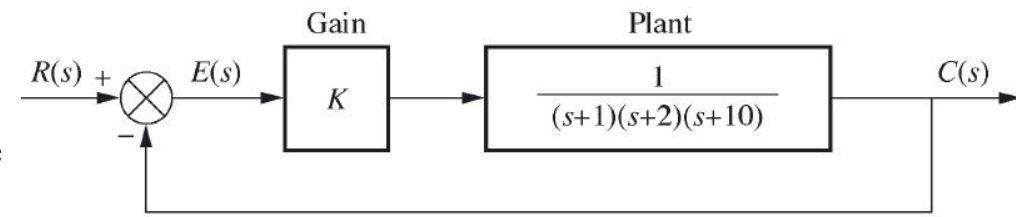


Figure (a)

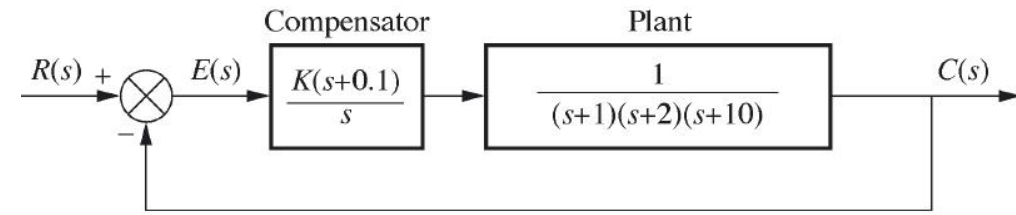


Figure (b)

For gain $K = 164.6$, searching along the line of $\zeta = 0.174$ for the *uncompensated system*: *dominant poles* are $0.694 \pm j3.926$ (third pole at -11.61) Figure (c).

This gain yields Position constant $K_p = \lim_{s \rightarrow 0} G(s) = \frac{164.6}{20} = 8.23$.

Hence, the steady-state error is: $e(\infty) = \frac{1}{1 + K_p} = \frac{1}{1 + 8.23} = 0.108$

We add an *ideal integral compensator* with a zero at -0.1 .

For gain $K = 158.2$, searching along the line of $\zeta = 0.174$ for the *compensated system*: *dominant poles* are $0.678 \pm j3.837$ (fourth pole at -0.0902) Figure (e).

Poles and gain are approximately the same \Rightarrow Same transient response

Damping Ratio unchanged (with $K = 158.2$). *Steady State Error is ZERO!*

Closed-loop system
a. before compensation;
b. after ideal integral compensation

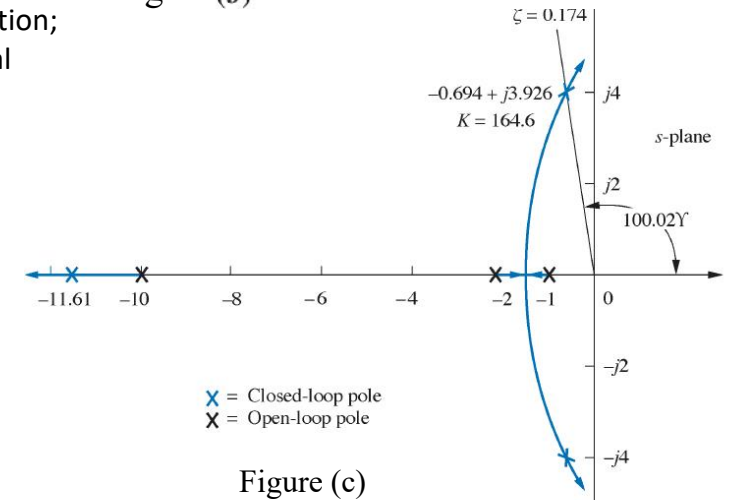


Figure (c)

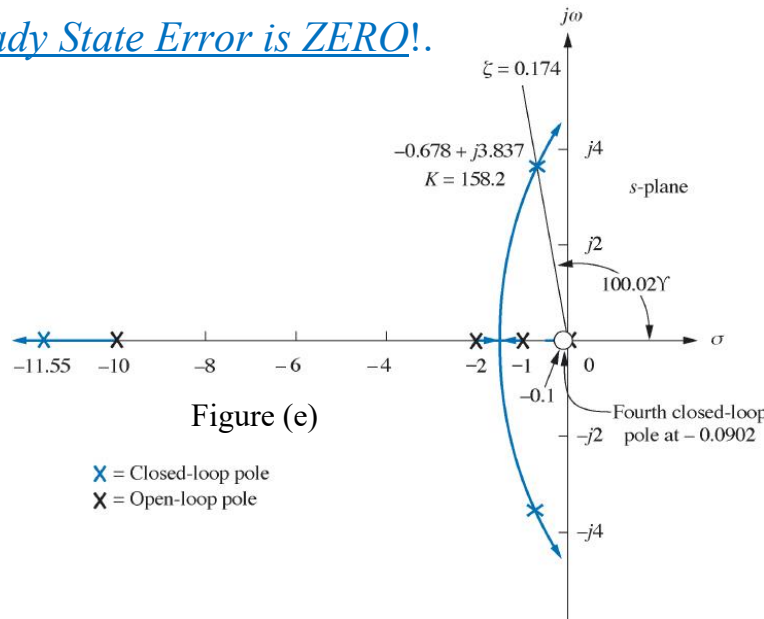


Figure (e)

\times = Closed-loop pole
 \times = Open-loop pole

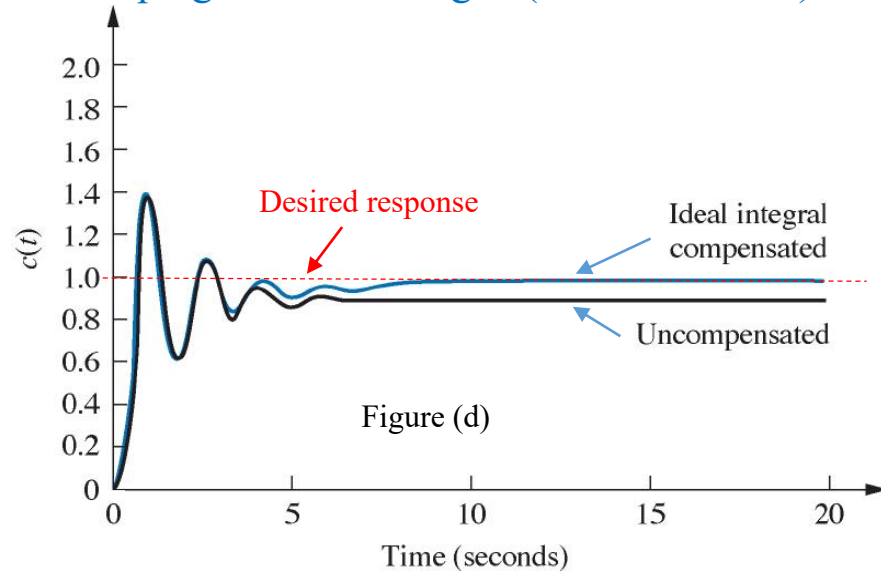


Figure (d)

Lag Compensation

Improving Steady-State Error

- Similar to the Ideal Integrator, however it has a pole not on the origin but close to the origin (fig c) due to the passive networks.

- Steady State Improvement:

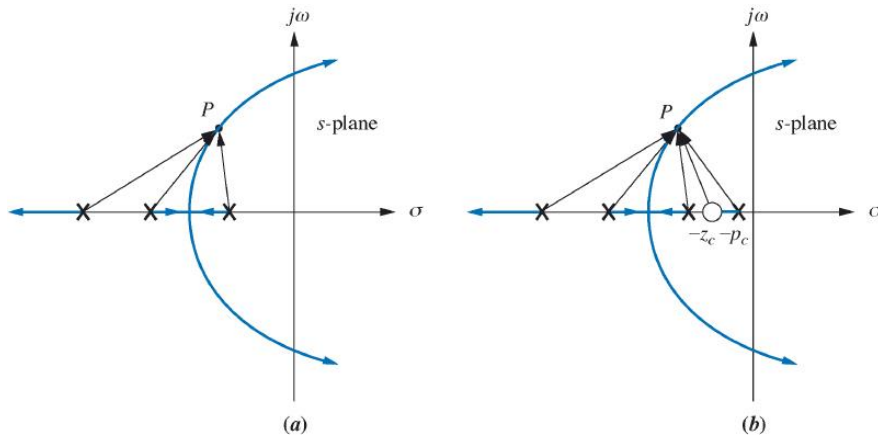
Before compensation: The static error constant, K_{v0} , for the system is:

$$K_{v0} = \lim_{s \rightarrow 0} sG(s) = K \frac{z_1 z_2 \dots}{p_1 p_2 \dots}$$

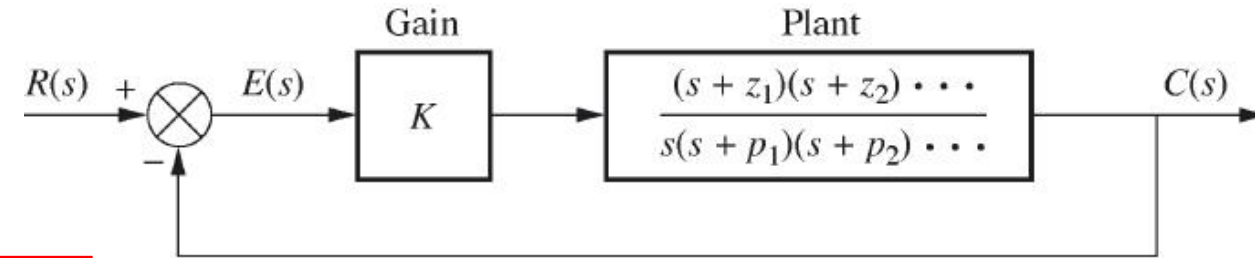
After compensation: $K_{v_{new}} = \frac{z_c}{p_c} K \frac{z_1 z_2 \dots}{p_1 p_2 \dots} \Rightarrow k_{v_{new}} = \frac{z_c}{p_c} \cdot k_{v_{new}}$

- The effect on the transient response is negligible:

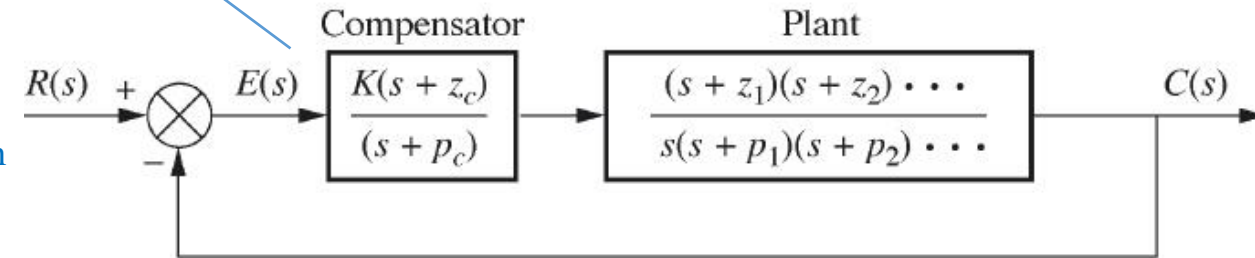
If the lag compensator pole and zero are close together, the angular contribution of the compensator to point P is approximately zero degrees. point P is still at approximately the same location on the compensated root locus.



Root locus: a. before lag compensation; b. after lag compensation

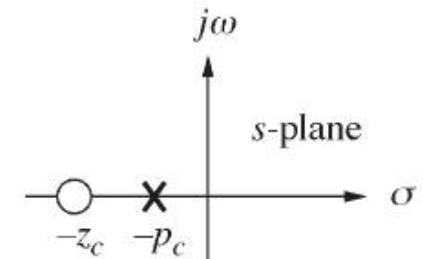


(a)



(b)

$$G_C(s) = \frac{(s + z_c)}{(s + p_c)}$$



(c)

Example2

Compensate the system of Figure (a), whose root locus is shown in Figure (b), to improve the steady-state error by a factor of 10 if the system is operating with a damping ratio of 0.174.

SOLUTION

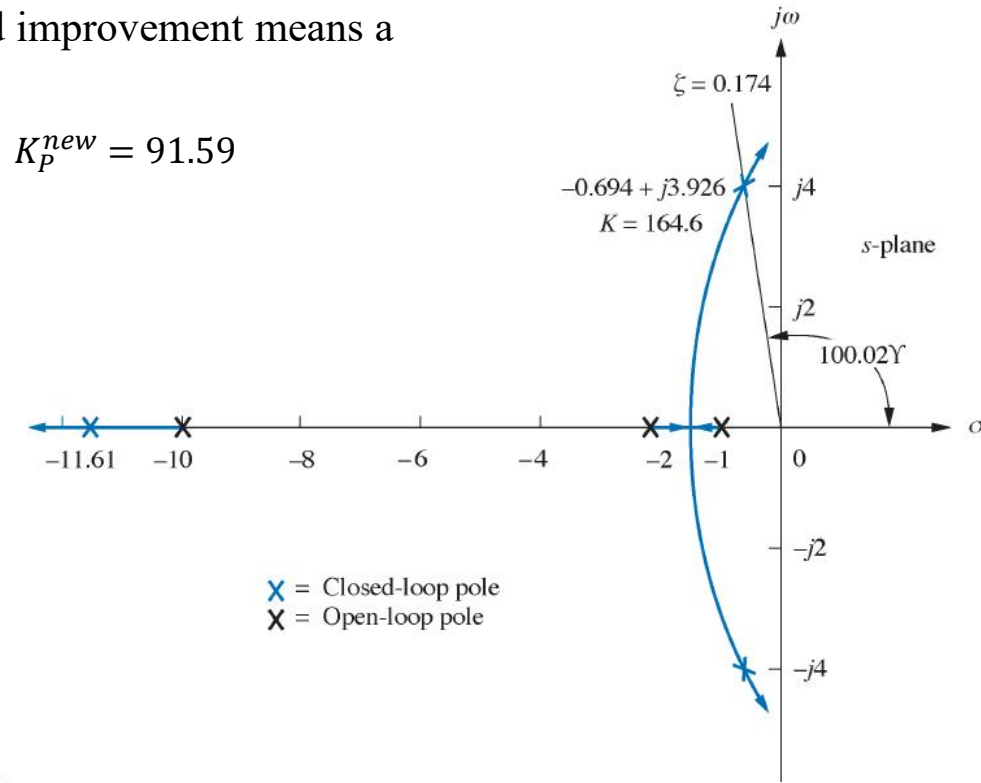
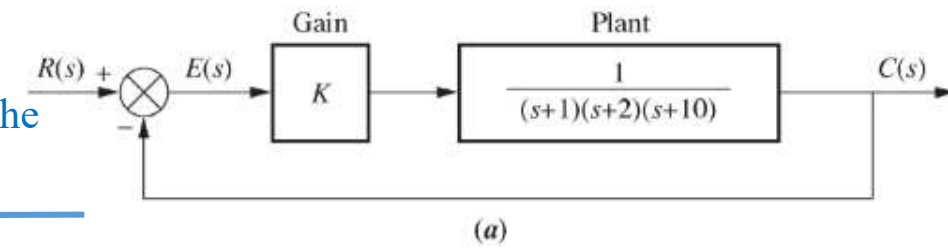
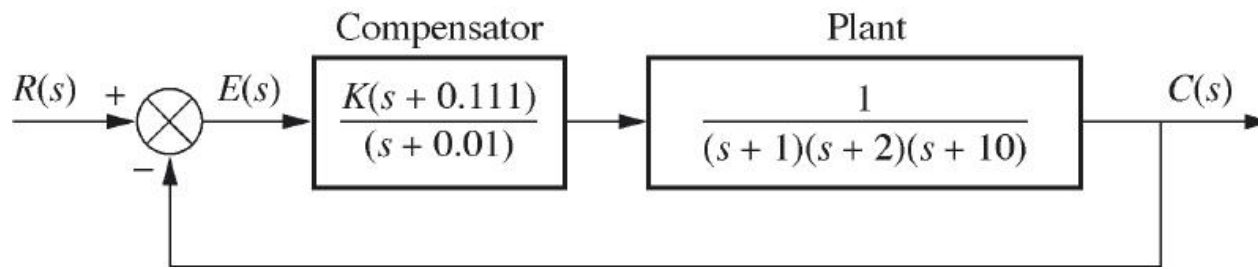
- From example 1: uncompensated system error was 0.108 with $K_p^{old} = 8.230$. A tenfold improvement means a steady-state error of:

$$e^{new}(\infty) = \frac{e^{old}(\infty)}{10} = \frac{0.108}{10} = 0.0108, \quad \text{since } e(\infty) = \frac{1}{1 + K_p^{new}} \Rightarrow K_p^{new} = 91.59$$

- For the compensated system $\frac{z_c}{p_c} = \frac{K_p^{new}}{K_p^{old}} = \frac{91.59}{8.23} = 11.13$

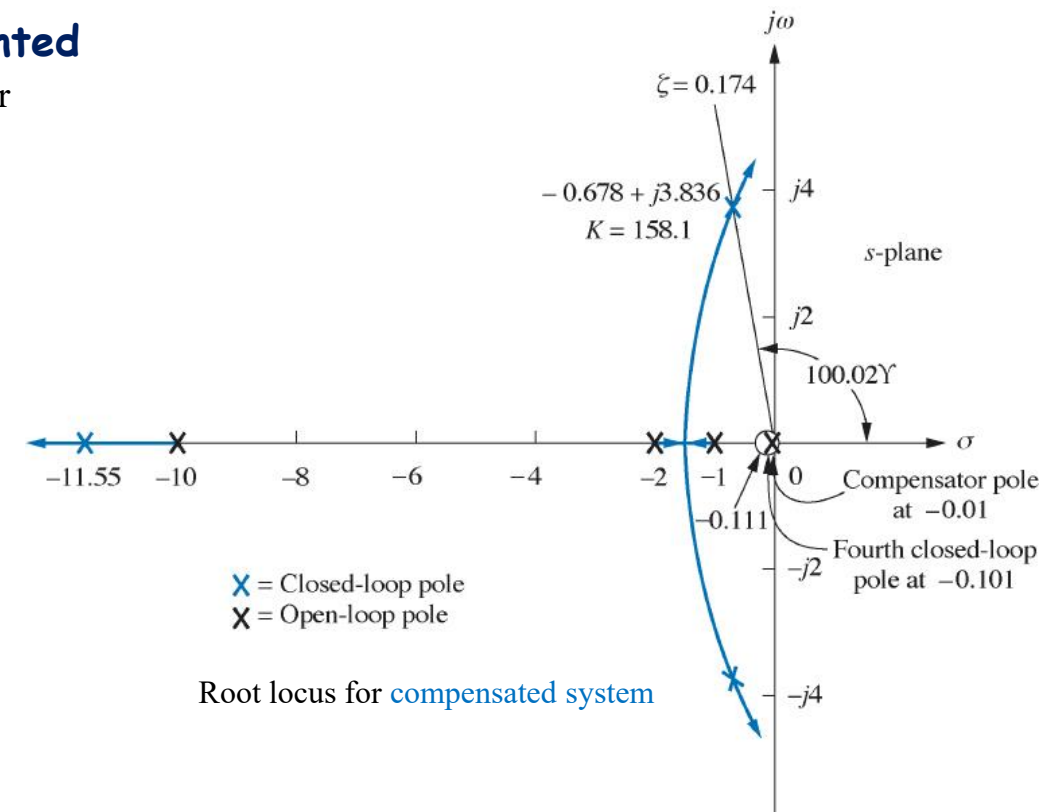
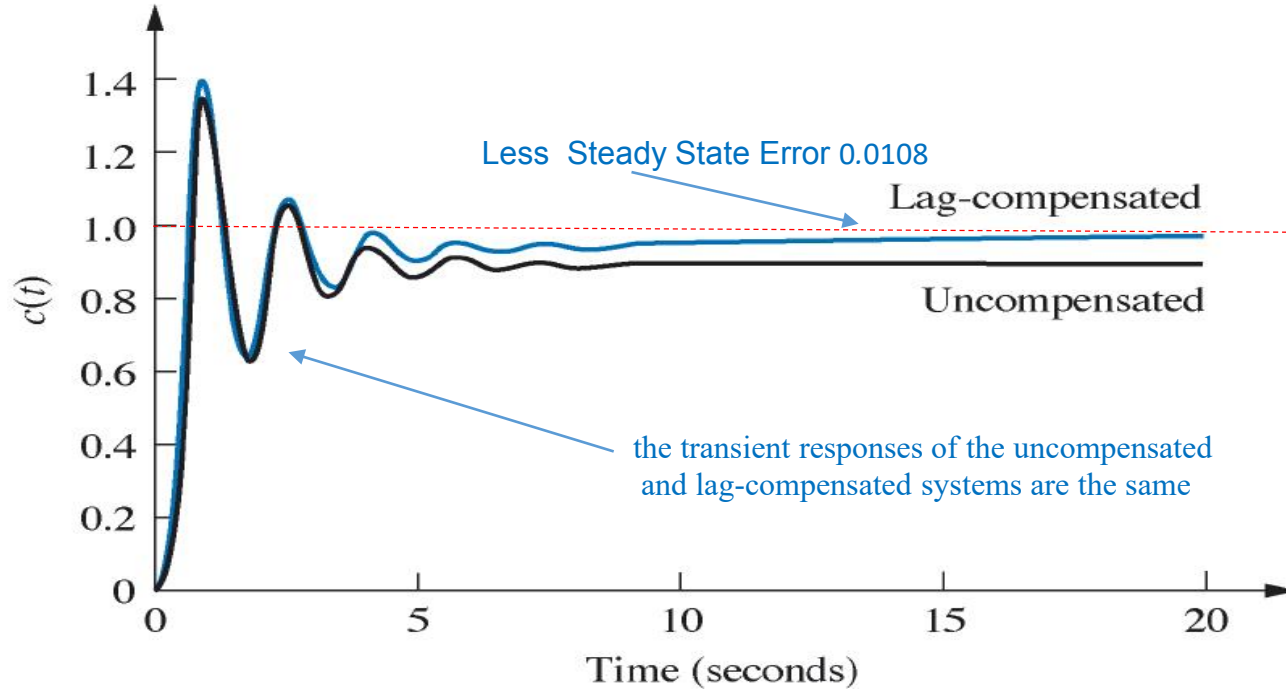
Arbitrarily selecting $p_c = 0.01 \Rightarrow z_c = 11.13 \cdot p_c = (11.13)(0.01)$
 $\Rightarrow z_c \approx 0.111$

- The compensated system



Example2-Conted

- The transient response of both systems is approximately the same with reduced steady state error



- Comparison of the Lag-Compensated and the Uncompensated Systems

Parameter	Uncompensated	Lag-compensated
Plant and compensator	$\frac{K}{(s+1)(s+2)(s+10)}$	$\frac{K(s+0.111)}{(s+1)(s+2)(s+10)(s+0.01)}$
K	164.6	158.1
K_p	8.23	87.75
$e(\infty)$	0.108	0.011
Dominant second-order poles	$-0.694 \pm j3.926$	$-0.678 \pm j3.836$
Third pole	-11.61	-11.55
Fourth pole	None	-0.101
Zero	None	-0.111

On the $\zeta = 0.174$ line: (compensated system):

The second-order dominant poles are at $-0.678 \pm j3.836$ ($K=158.1$)

The third and fourth closed-loop poles are at -11.55 and -0.101.

The fourth pole of the compensated system cancels its zero.


Ideal Derivative Compensation (PD)

Improving Transient Response

- The objective is to design a response that has a desirable percent overshoot and a shorter settling time than the uncompensated system. (two approaches).

1. **Ideal derivative compensation (Proportional-plus-Derivative (PD) active elements)**: a pure differentiator is added to the forward path of the feedback control system. $G_1(s) = s + z_c$ sensitive to high frequency noise.

2. **Lead Compensation**: (not pure differentiation) approximates differentiation with a passive network by adding a zero and a more distant pole to the forward-path transfer function.

$$G_1(s) = K \frac{s + z_c}{s + p_c}$$


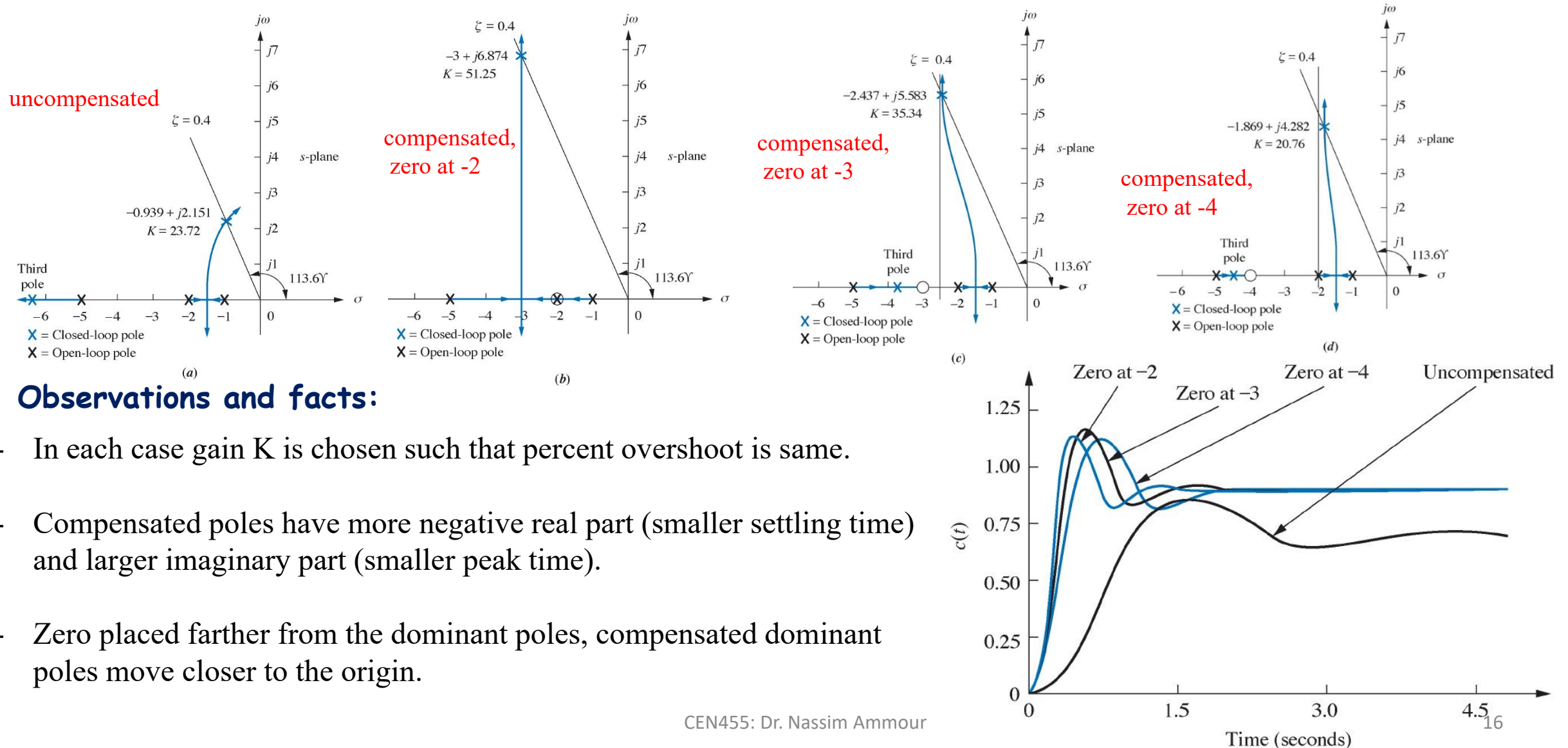
Less sensitive to high frequency noise.

- The transient response of a system can be selected by choosing an appropriate closed-loop pole location on the s-plane.
- If this point is on the root locus, then a simple gain adjustment is all that is required in order to meet the transient response specification.
- If the closed loop root locus doesn't go through the desired point, it needs to be reshaped (add poles and zeros in the forward path).
- One way to speed up the original system is to add a single zero to the forward path. $G_c(s) = s + z_c$

Ideal Derivative Compensation (PD)

Improving Transient Response

- See how it affects by an example of a system operating with a damping ratio of 0.4:

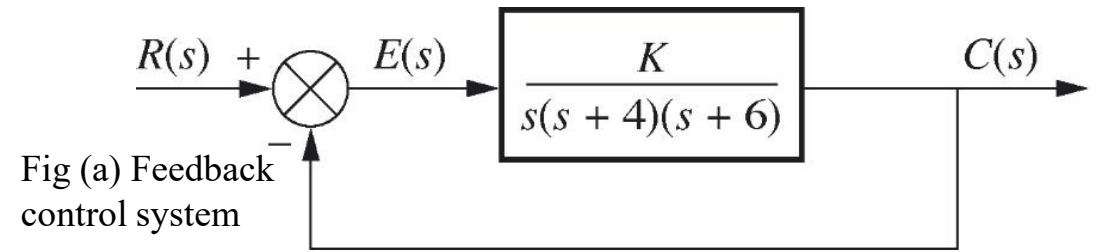


Observations and facts:

- In each case gain K is chosen such that percent overshoot is same.
- Compensated poles have more negative real part (smaller settling time) and larger imaginary part (smaller peak time).
- Zero placed farther from the dominant poles, compensated dominant poles move closer to the origin.

Example 3₁

Given the system of Figure (a), design an ideal derivative compensator to yield a 16% overshoot, with a threefold reduction in settling time.



SOLUTION

The performance of the *uncompensated system* operating with 16% overshoot fig (b).

16% Overshoot $\Rightarrow \zeta = 0.504$, $\xrightarrow[\text{Dominant second-order poles}]{\text{along damping ratio line}}$ $-1,205 \pm j2.064$.
(with $k = 43.35$ and third pole at -7.59 .)

Settling time $\Rightarrow T_s = \frac{4}{\zeta\omega_n} = \frac{4}{1.205} = 3.320$

Location of the *compensated system's* dominant poles. (Desired poles)

$T_s^{new} = \frac{T_s^{old}}{3} = 1.107$ $\xrightarrow{\text{threefold reduction in the settling time}}$

$\Rightarrow \sigma = \frac{4}{T_s^{new}} = \frac{4}{1.107} = 3.613$ $\xrightarrow{\text{real part of the compensated system's dominant, second-order pole}}$

$\omega_d = 3.613 \tan(180^\circ - 120.26^\circ) = 6.193$ $\xrightarrow{\text{Imaginary part of the compensated system's dominant pole on line } \zeta = 0.504}$

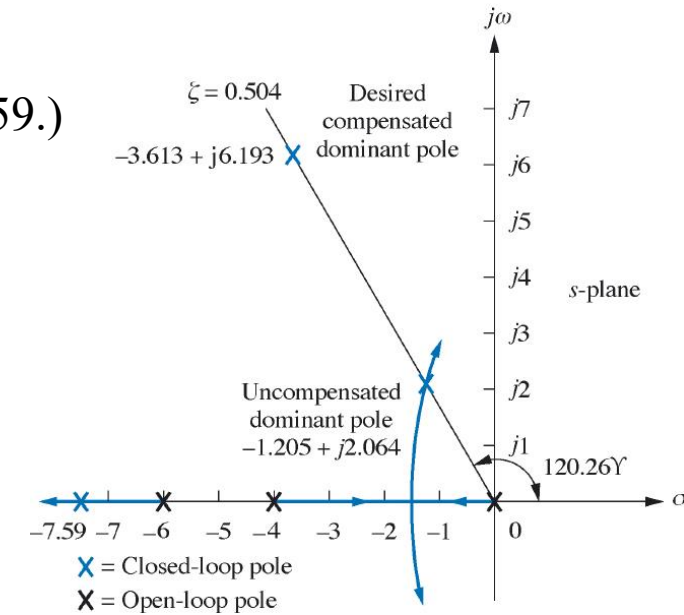


Fig (b) Compensated dominant pole

Example 3₂

Zero contribution angle $> 90^\circ \rightarrow$ zero position less than desired pole real part.

Design the location of the compensator zero

- The angle contribution of poles for the desired pole location: -275.6° .
- To achieve -180° the angle contribution of the placed zero should be: $-275.6^\circ + x = -180^\circ \rightarrow x = 95.6^\circ$
- From the fig (c):

$$\frac{6.193}{3.613 - \sigma} = \tan(180 - 95.6) \Rightarrow \sigma = 3.006$$

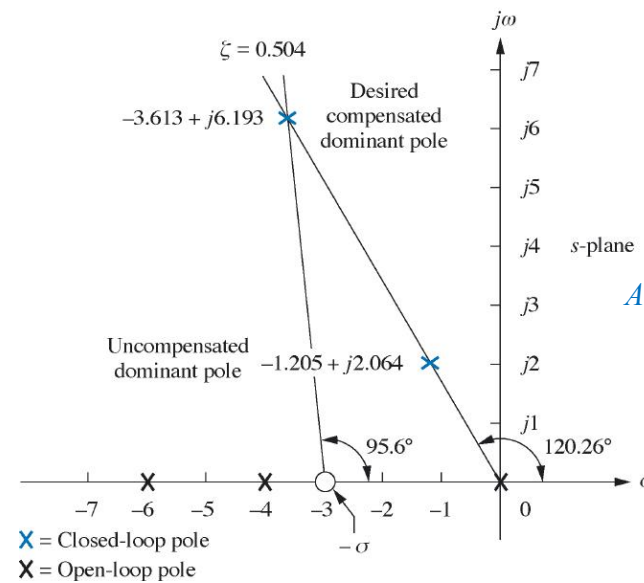
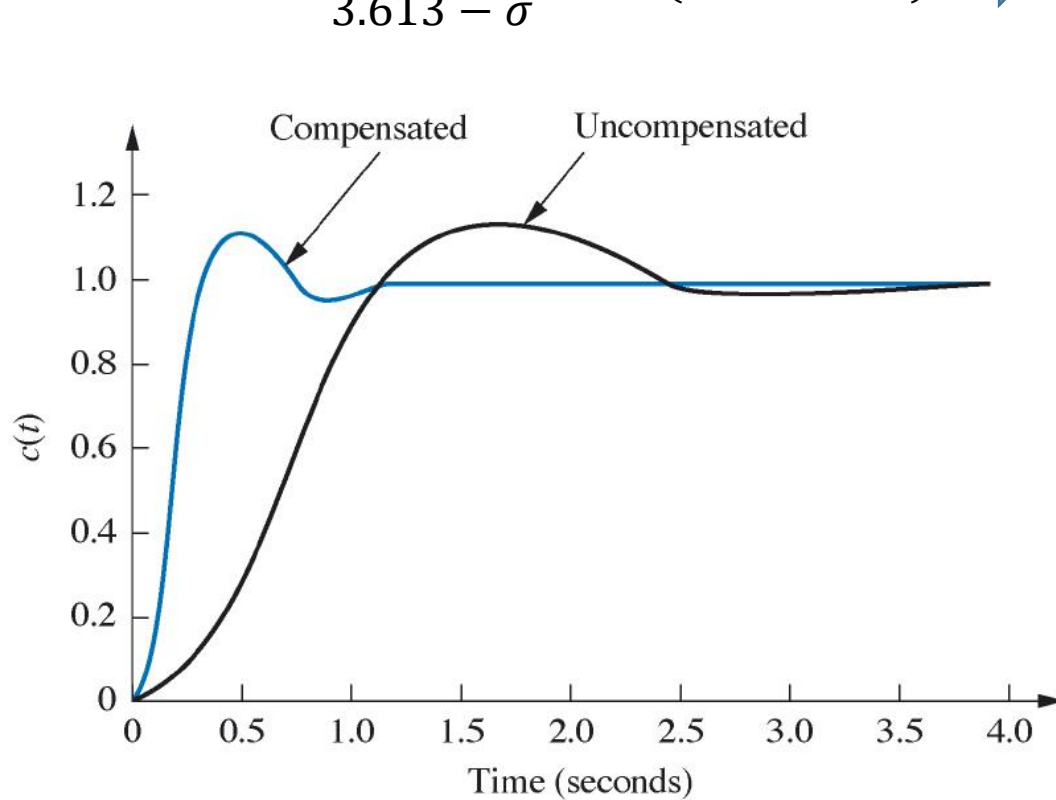


Fig (c) Evaluating the location of the compensating zero

Adding zero

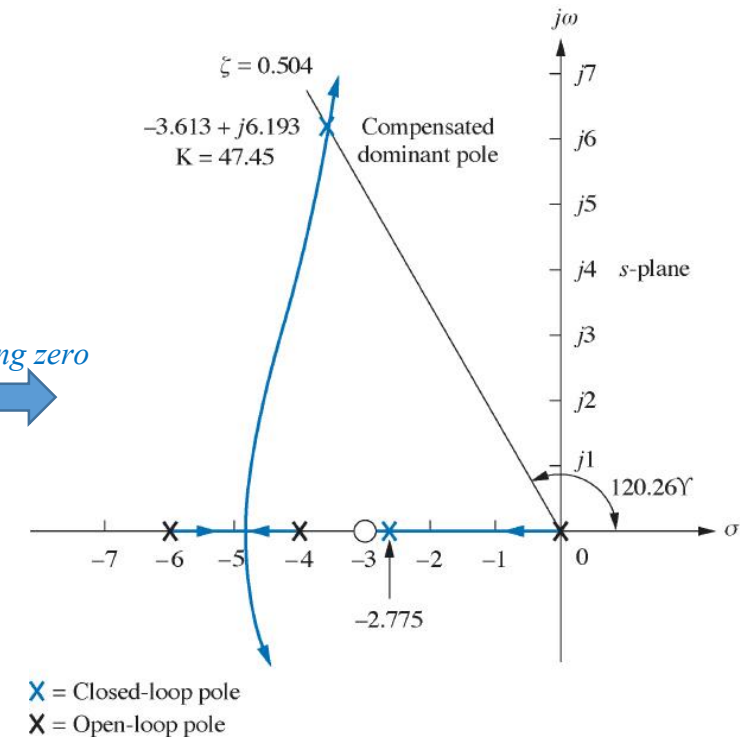


Fig (c) Root locus for the compensated system

Fig (d) Uncompensated and compensated system step responses

Lead Compensation

Basic Idea: The difference between 180° and the sum of the angles must be the angular contribution required of the compensator.

Example: looking at the Figure, we see that:

$$\theta_2 - \theta_1 - \theta_3 - \theta_4 + \theta_5 = (2k + 1)180^\circ$$

where $\theta_1 - \theta_2 = \theta_c$ is the angular contribution of the compensator

- There are infinitely many choices of z_c and p_c providing same θ_c

Example 4₁

Design three lead compensators for the system in Figure to reduce the settling time by a factor of 2 while maintaining 30% overshoot.

SOLUTION

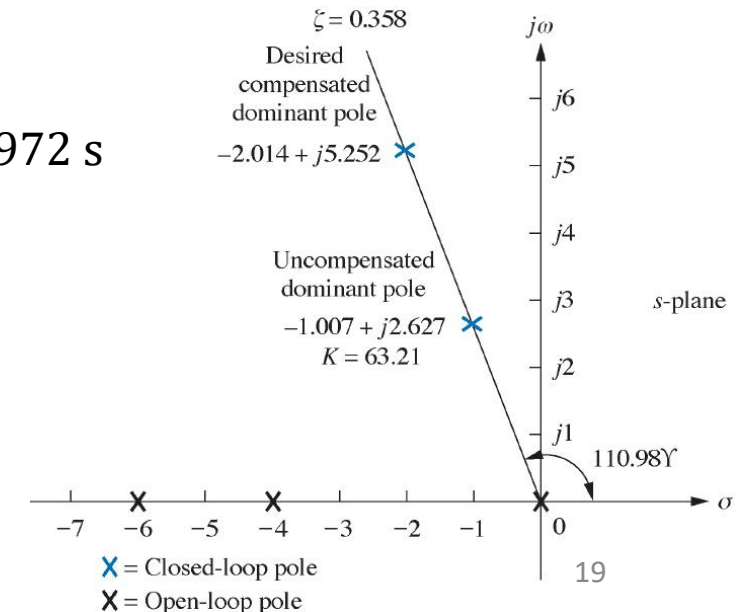
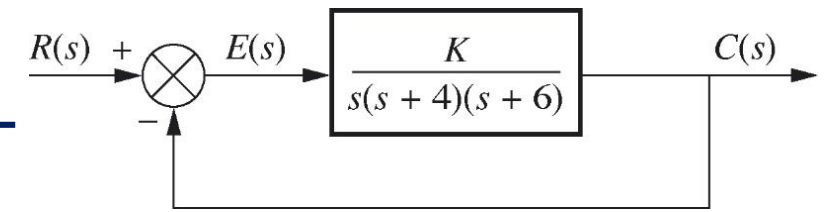
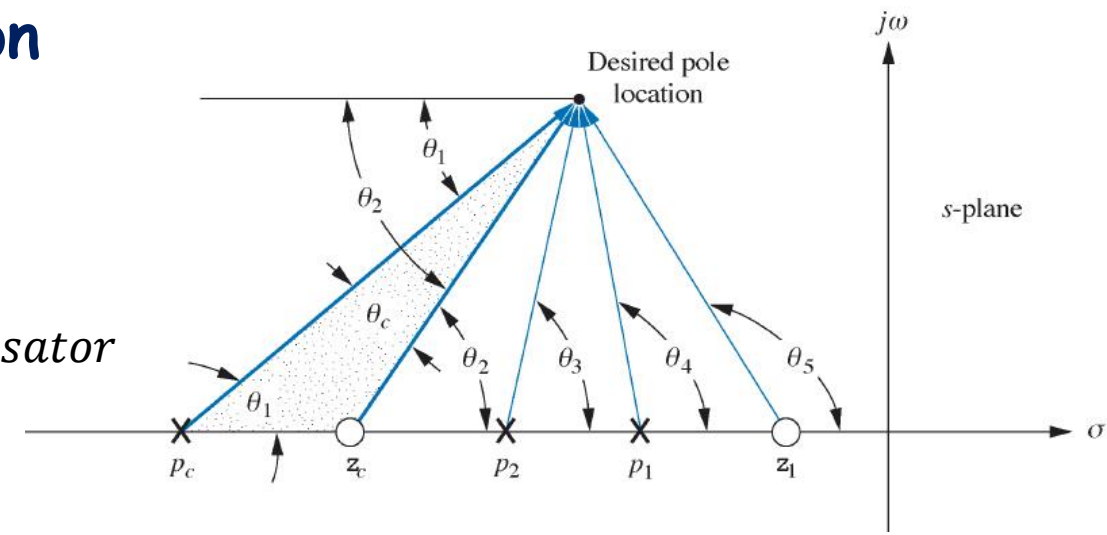
- Characteristics of the uncompensated system operating at 30% overshoot

30% Overshoot $\xrightarrow{\text{damping ratio}}$ $\zeta = 0.358$, $\xrightarrow{\text{Dominant second-order pair of poles along damping ratio line}}$ $-1,007 \pm j2.627$. $\xrightarrow{\text{settling time From pole's real part}}$ $T_s = 4/1.007 = 3.972 \text{ s}$

- Design point (*Desired Poles location*)

twofold reduction in settling time $\xrightarrow{\quad} T_s = 3.972/2 = 1.986 \text{ s}$ $\xrightarrow{\text{real part of the desired pole location}}$ $-\zeta\omega_n = -4/T_s = -2.014$

Imaginary part of the desired pole location $\xrightarrow{\quad} \omega_d = -2.014 \tan(110.98^\circ) = 5.252$



Example 4₂

- Lead compensator Design.

Place the zero on real axis at -5 (arbitrarily as possible solution).
sum the angles (this zero and uncompensated system's poles and zeros),

resulting angle $\theta_0 = -172.69^\circ$ $\xrightarrow{\text{the angular contribution required from the compensator pole}}$ $\theta_c = -180^\circ + 172.69^\circ = -7.31^\circ$

location of the compensator pole $\xrightarrow{\text{From geometry in fig(a)}}$ $\frac{5.252}{p_c - 2.014} = \tan(7.31^\circ) \xrightarrow{\text{compensator pole}}$ $p_c = 42.96$

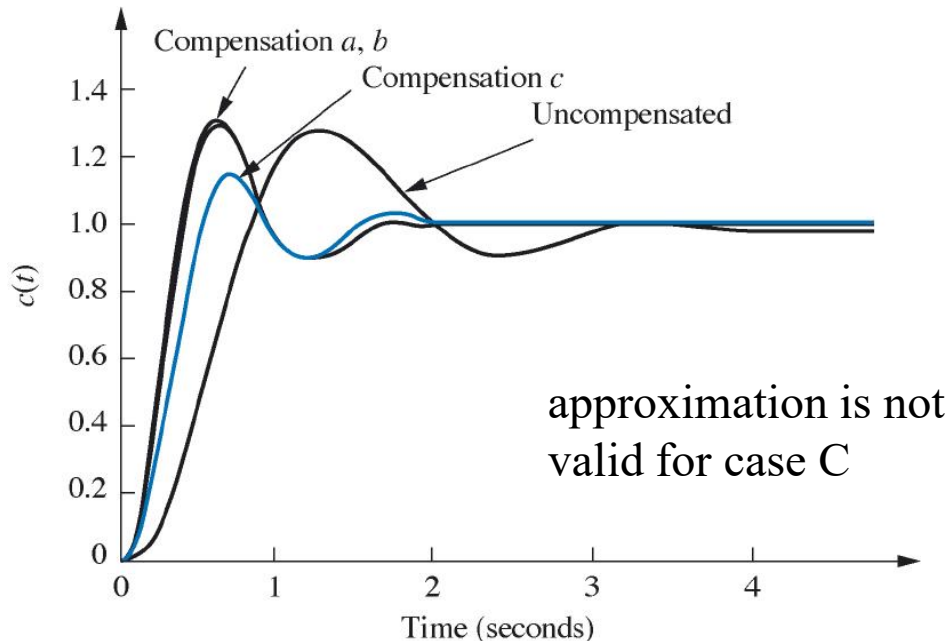
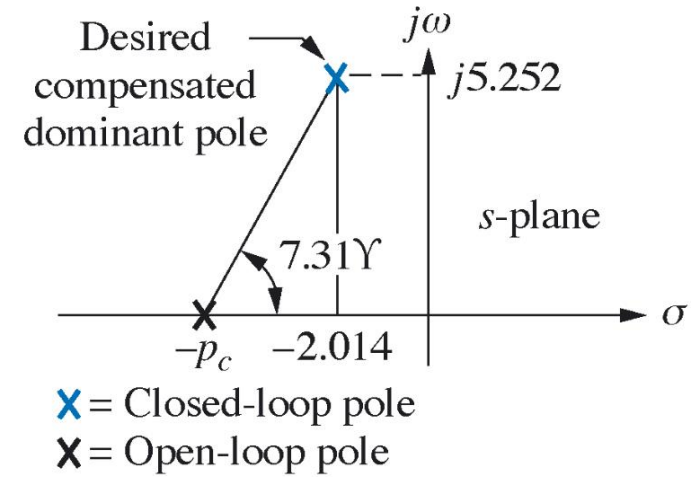
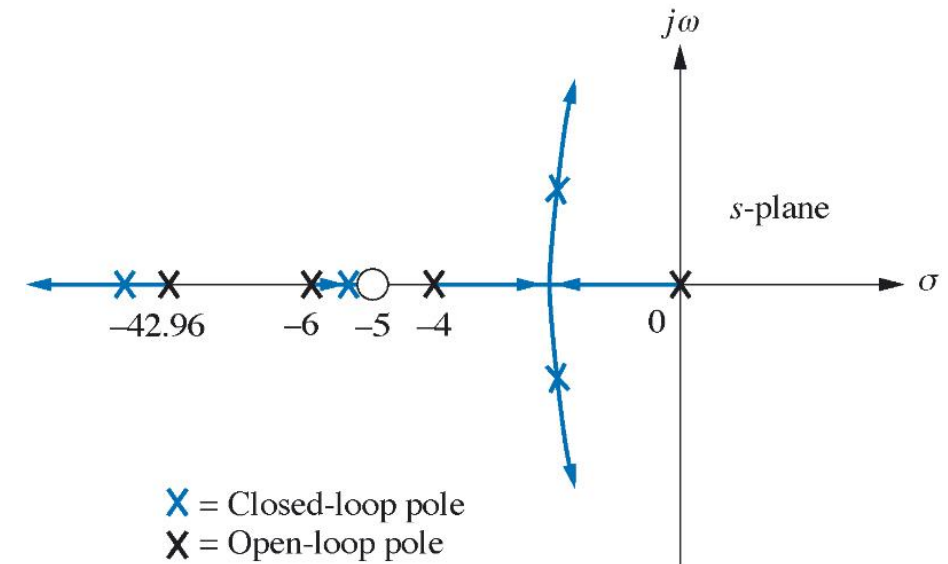


Fig (c) Uncompensated system and lead compensation responses (zeros at a:-5, b:-4 c: -2)



Note: This figure is not drawn to scale.

Fig (a) s-plane picture used to calculate the location of the compensator pole



Note: This figure is not drawn to scale.

Fig (b) Compensated system root locus

Improving Steady-State Error and Transient Response

- Combine the design techniques to obtain improvement in steady-state error and transient response *independently*.
 - First improve the transient response.(PD or lead compensation).
 - Then improve the steady-state response. (PI or lag compensation).
- Two Alternatives
 - PID (Proportional-plus-Integral-plus-Derivative) (with Active Elements).
 - Lag-Lead Compensator. (with Passive Elements).

PID Controller Design

- Transfer Function of the compensator (two zeros and one pole):

$$G_c(s) = k_1 + \frac{k_2}{s} + k_3 s = \frac{k_1 s + k_2 + k_3 s^2}{s} = \frac{k_3(s^2 + \frac{k_1}{k_3}s + \frac{k_2}{k_3})}{s}$$

- Design Procedure (Fig (a))

1. From the requirements figure out the desired pole location to meet transient response specifications.
2. Design the PD controller to meet transient response specifications.
3. Check validity (all requirements have been met) of the design by simulation.
4. Design the PI controller to yield the required steady-state error.
5. Determine the gains, k_1, k_2 and k_3 (Combine PD and PI).
6. Simulate the system to be sure all requirements have been met.
7. Redesign if simulation shows that requirements have not been met.

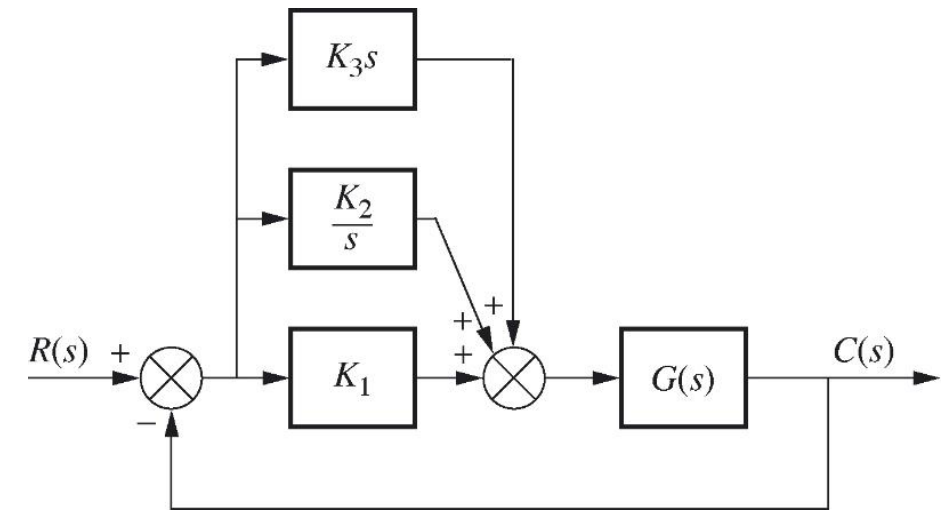


Fig (a) PID controller implementation

Example 5₁

Given the system of Figure (a), design a **PID** controller so that the system can operate with a **peak time that is two-thirds** that of the uncompensated system at **20% overshoot** and with **zero steady-state error** for a step input.

SOLUTION

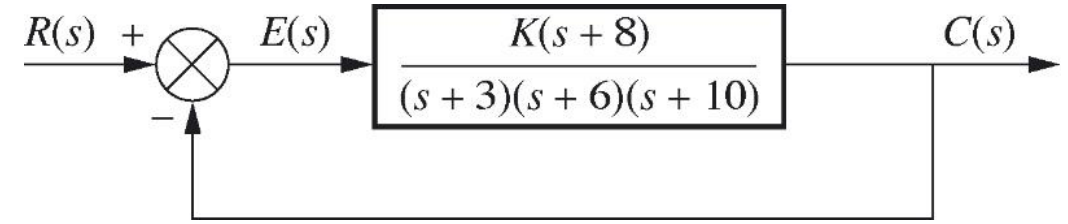


Fig (a) Uncompensated feedback control system

- Evaluation of the uncompensated system

20% overshoot $\xrightarrow{\text{damping ratio}}$ $\zeta = 0.456$, $\xrightarrow{\text{Dominant second-order pair of poles}}$ $-5.415 \pm j10.57$ with gain of 121.5.

$\xrightarrow{\text{Peak time}}$ $T_p = \frac{\pi}{\omega_d} = 0.297 \text{ seconds}$

between -8 and -10 for a gain equivalent to that at the dominant poles

A third pole at -8.169

- To reduce the peak time to two-thirds. (find the compensated system's dominant pole location)

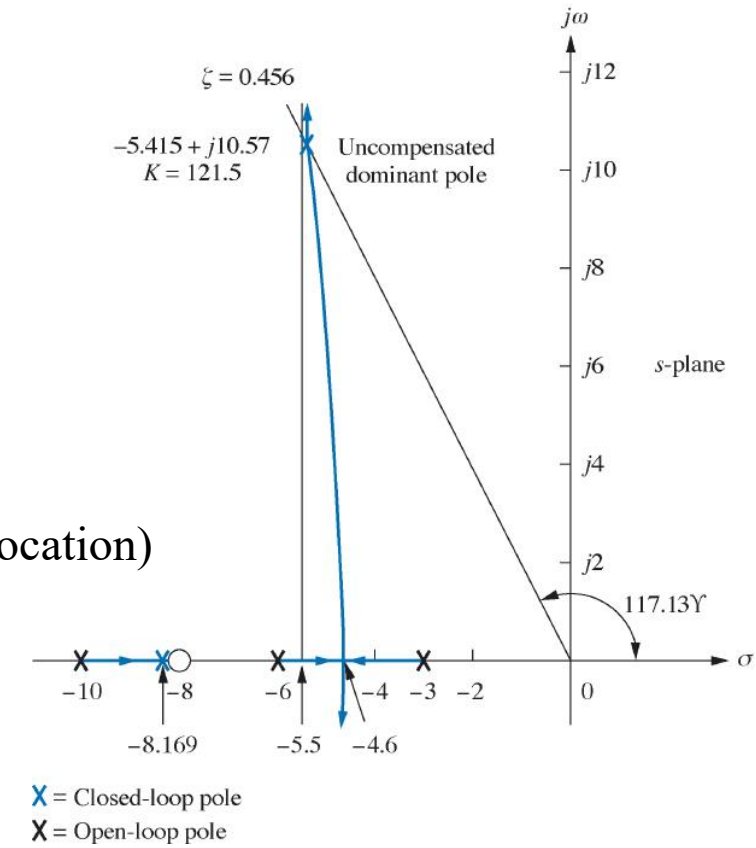
The imaginary part

$$\omega_d = \frac{\pi}{T_p} = \frac{\pi}{(2/3)(0.297)} = 15.87$$

The real part

$$\sigma = \frac{\omega_d}{\tan(117.13^\circ)} = -8.13$$

$$\tan(117.13^\circ) = -\tan(180 - 117.13^\circ)$$



Example 5₂

- Design of the compensator

(sum of angles from the uncompensated system's poles and zeros to the desired compensated dominant pole is -198.37°)

the required contribution
from the compensator zero z_c

$$-198.37^\circ + \theta_c = -180^\circ \Rightarrow \theta_c = 18.37^\circ$$

compensating
zero's location.

From geometry
in Fig(a)

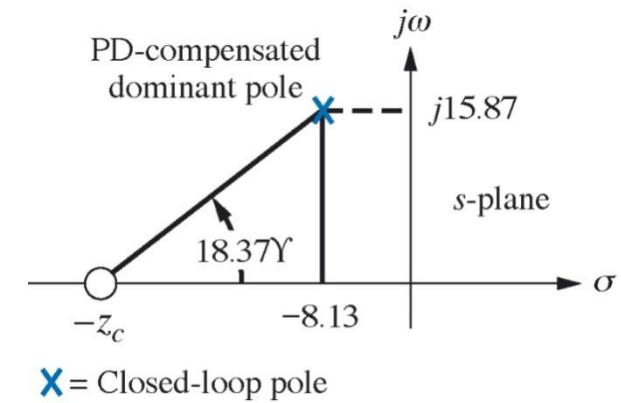
$$\frac{15.87}{z_c - 8.13} = \tan 18.37^\circ$$

the PD controller.

$$z_c = 55.92 \Rightarrow G_{PD}(s) = (s + 55.92)$$

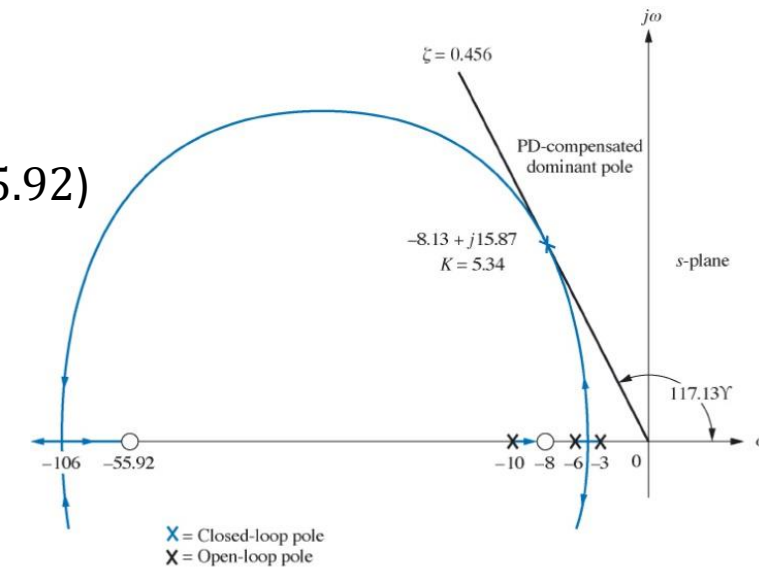
gain at the design point

$$k = 5.34$$



Note: This figure is not drawn to scale.

Fig (a) Calculating the PD compensator zero



Note: This figure is not drawn to scale.

Fig (b) Root locus for PD-compensated system

- The PD-compensated system is simulated. Fig (b) (next slide) shows the reduction in peak time and the improvement in steady-state error over the uncompensated system. (step 3 and 4)

Example 5₃

- A PI controller is used to reduce the steady-state error to zero (for PI controller the zero is placed at -0.5 close to the origin)

PI controller is used as

$$G_{PI}(s) = \frac{s + 0.5}{s}$$

Searching the 0.456 damping ratio line

we find the dominant, second-order poles

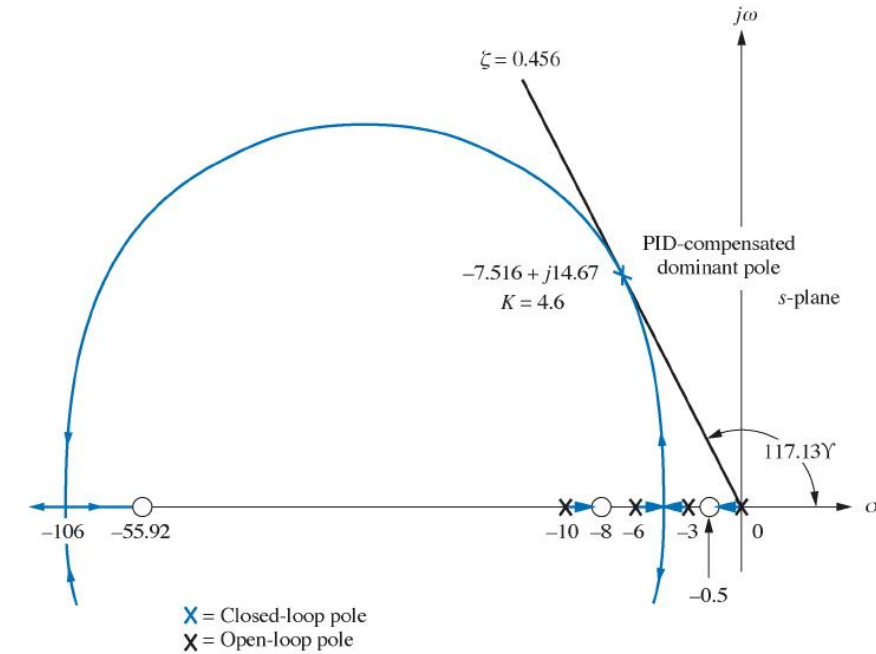
$$-7.516 \pm j 14.67 \text{ with associated gain } k = 4.6$$

- Now we determine the gains (the PID parameters),

$$G_{PID}(s) = \frac{k(s + 55.92)(s + 0.5)}{s} = \frac{4.6(s + 55.92)(s + 0.5)}{s}$$

$$= 256.5 + 128.6 \frac{1}{s} + 4.6 s = k_1 + k_2 \frac{1}{s} + k_3 s$$

Matching: $k_1 = 256.5$, $k_2 = 128.6$, $k_3 = 4.6$



Note: This figure is not drawn to scale.

Fig (a) Root locus for PID-compensated system

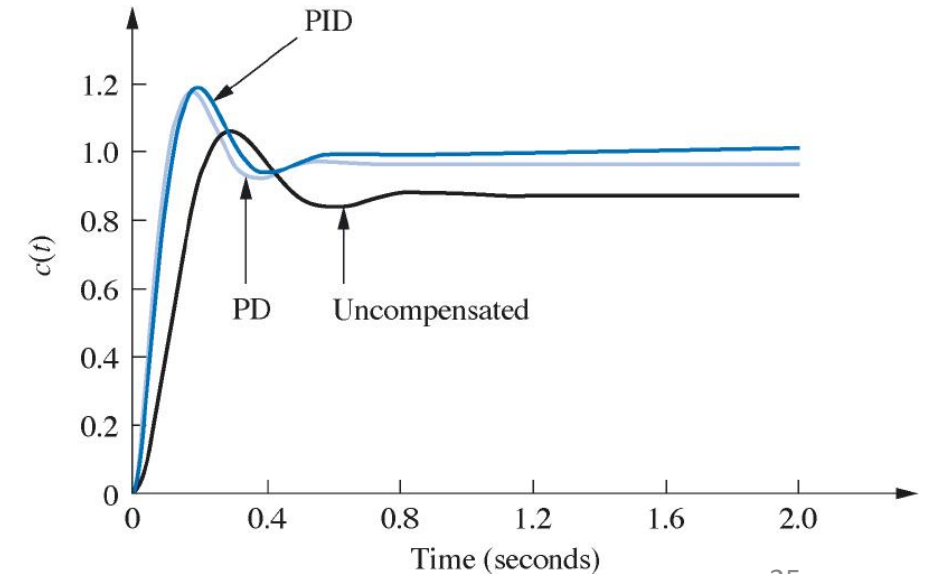


Fig (b) Step responses for uncompensated, PD compensated, and PID compensated systems

Lag-Lead Compensator Design

(Cheaper solution than PID)

- First design the lead compensator to improve the transient response. Next we design the lag compensator to meet the steady-state error requirement.
- Design procedure:
 1. Determine the desired pole location based on specifications. (Evaluate the performance of the uncompensated system).
 2. Design the lead compensator to meet the transient response specifications.(zero location, pole location, and the loop gain).
 3. Evaluate the steady state performance of the lead compensated system to figure out required improvement.(simulation).
 4. Design the lag compensator to satisfy the improvement in steady state performance.
 5. Simulate the system to be sure all requirements have been met. (If not met redesign)

Example 6₁

Design a lag-lead compensator for the system of Figure so that the system will operate with 20% overshoot and a twofold reduction in settling time. Further, the compensated system will exhibit a tenfold improvement in steady-state error for a ramp input.

SOLUTION

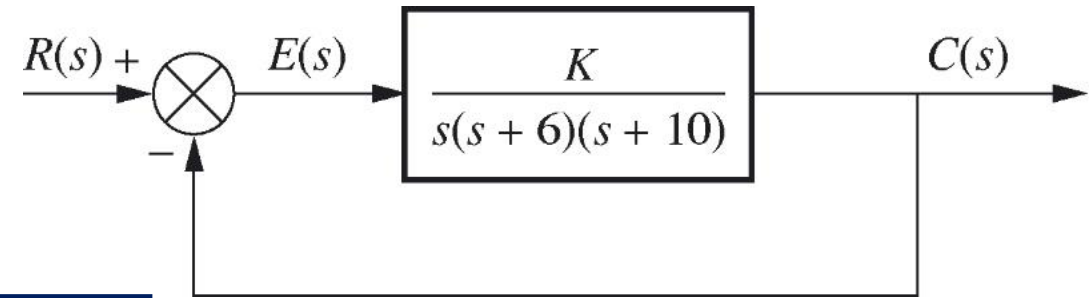


Fig (a) Uncompensated system

- Step 1:** Evaluation of the uncompensated system

20% Overshoot $\xrightarrow{\text{damping ratio}}$ $\zeta = 0.456$, $\xrightarrow[\text{along damping ratio line}]{\text{Dominant second-order pair of poles}}$ $-1,794 \pm j3.501$ with gain of 192.1.

- Step 2 :** Lead compensator design (selection of the location of the compensated system's dominant poles).

Twofold reduction of settling time $\xrightarrow[\text{the real part of the dominant pole}]{\text{the imaginary part of the dominant pole}}$ $-\zeta \omega_n = -2(1.794) = -3.588$ $\xrightarrow{\text{the imaginary part of the dominant pole}}$ $\omega_d = \zeta \omega_n \tan(117.13^\circ) = 7.003$

lead compensator design. $\xrightarrow[\text{Arbitrarily select a location for the lead compensator zero.}]{\text{the real part of the dominant pole}}$ $z_c = -6$

- compensator zero coincident with the open-loop pole to eliminate a zero and leave the lead-compensated system with three poles. (same number that the uncompensated system has)

Example 6₂

- Finding the location of the compensator pole.
- Sum the angles to the design point from the uncompensated system's poles and zeros and the compensator zero and get -164.65° .
- The difference between 180° and this quantity is the angular contribution required from the compensator pole (-15.35°).
- Using the geometry shown in Figure (b)

Compensator pole. $\longrightarrow \frac{7.003}{p_c - 3.588} = \tan(15.35^\circ) \longrightarrow p_c = -29.1$

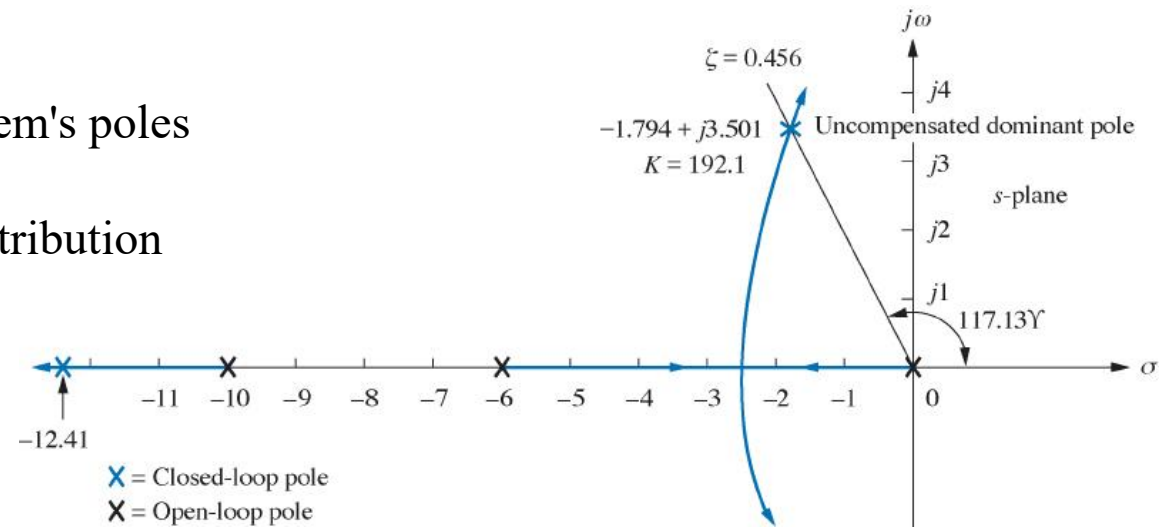


Fig (a) Root locus for uncompensated system

- The complete root locus for the lead-compensated system is sketched in Figure (c)

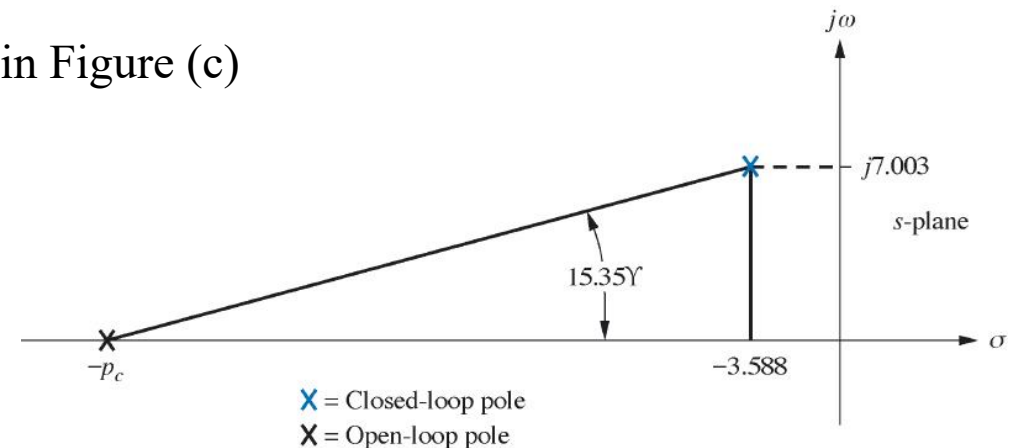


Fig (b) Evaluating the compensator pole

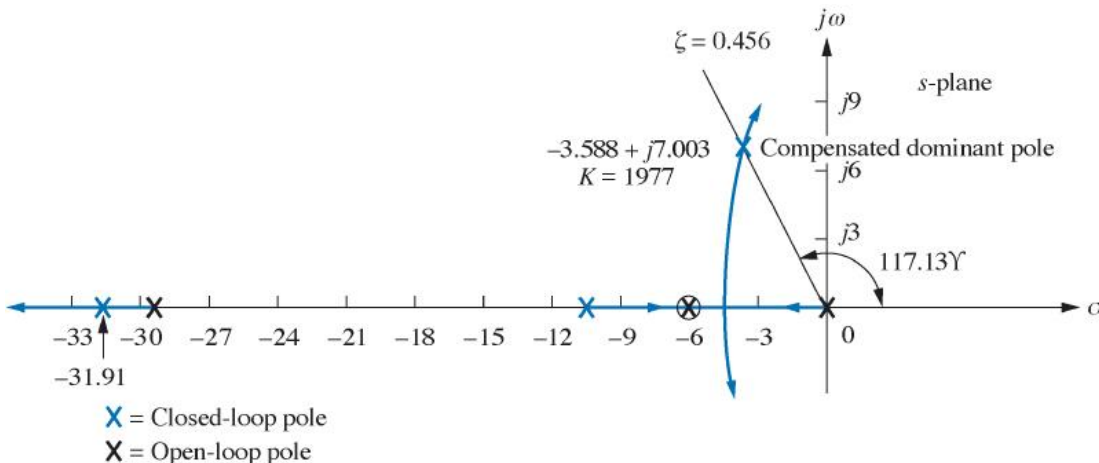
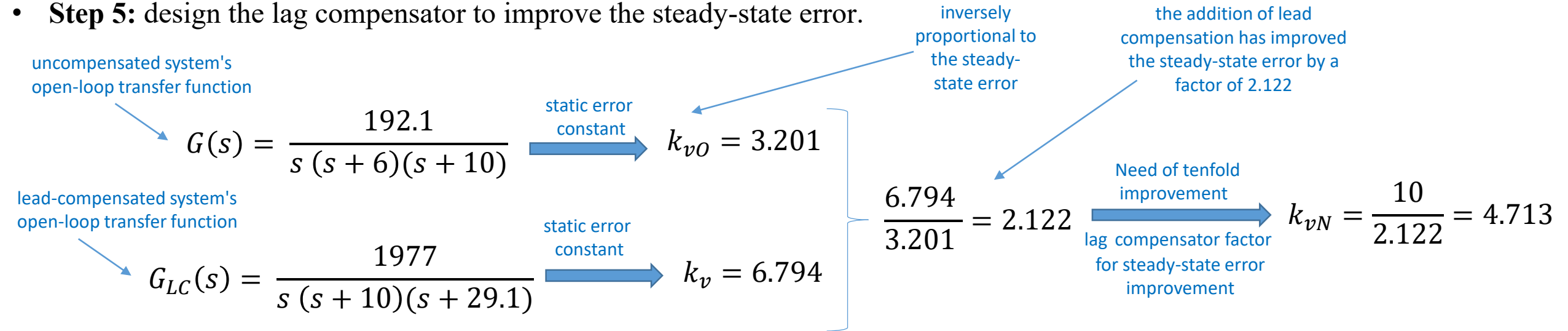


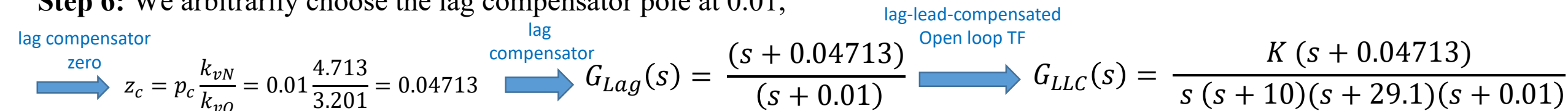
Fig (c) Root locus for lead-compensated system

Example6₃

- **Steps 3 and 4:** Check the design with a simulation. (The result for the lead compensated system is shown in Figure(a) and is satisfactory.)
- **Step 5:** design the lag compensator to improve the steady-state error.



Step 6: We arbitrarily choose the lag compensator pole at 0.01,



- The uncompensated system pole at -6 canceled the lead compensator zero at -6.
- Drawing the complete root locus for the lag-lead-compensated system and by searching along the 0.456 damping ratio line

closed-loop dominant poles $\rightarrow p_c = -3.574 \pm j 6.976$ with a gain of 1971.

Example 6₄

- Fig (b) shows the complete draw of the lag-lead-compensated root locus.
- The lag-lead compensation has indeed increased the speed of the system (settling time or the peak time).

Step 7: The final proof of our designs is shown by the simulations of Figure (b)

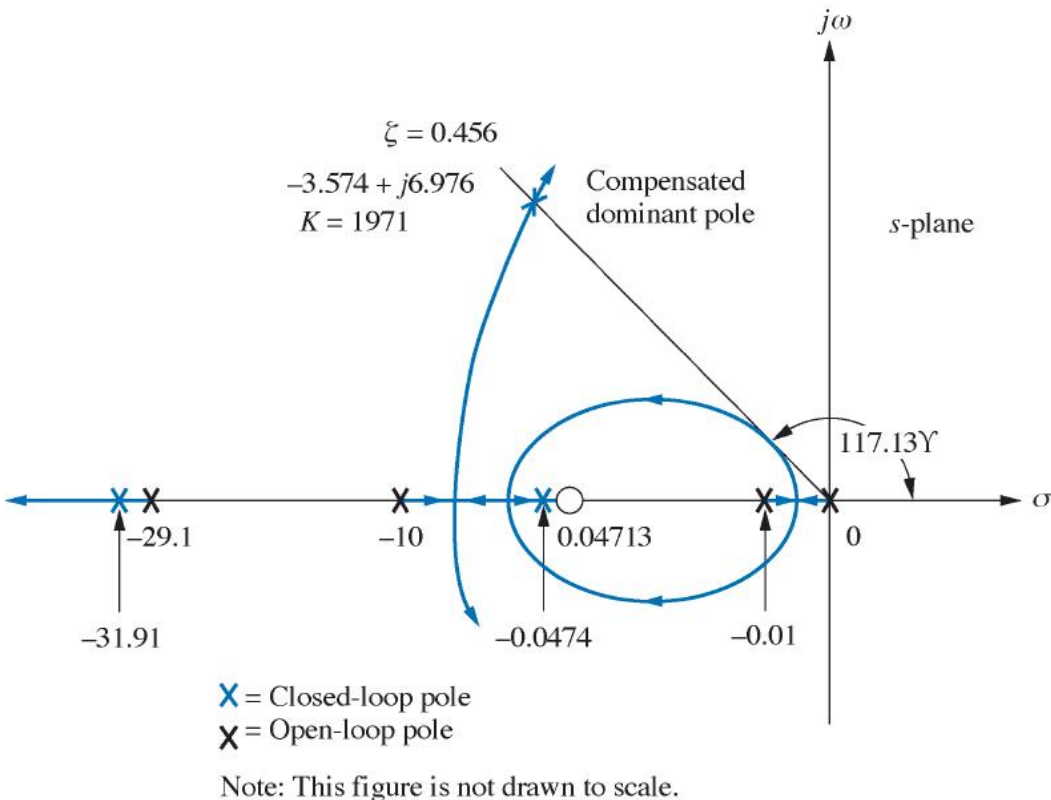


Fig (a) Root locus for lag-lead-compensated system

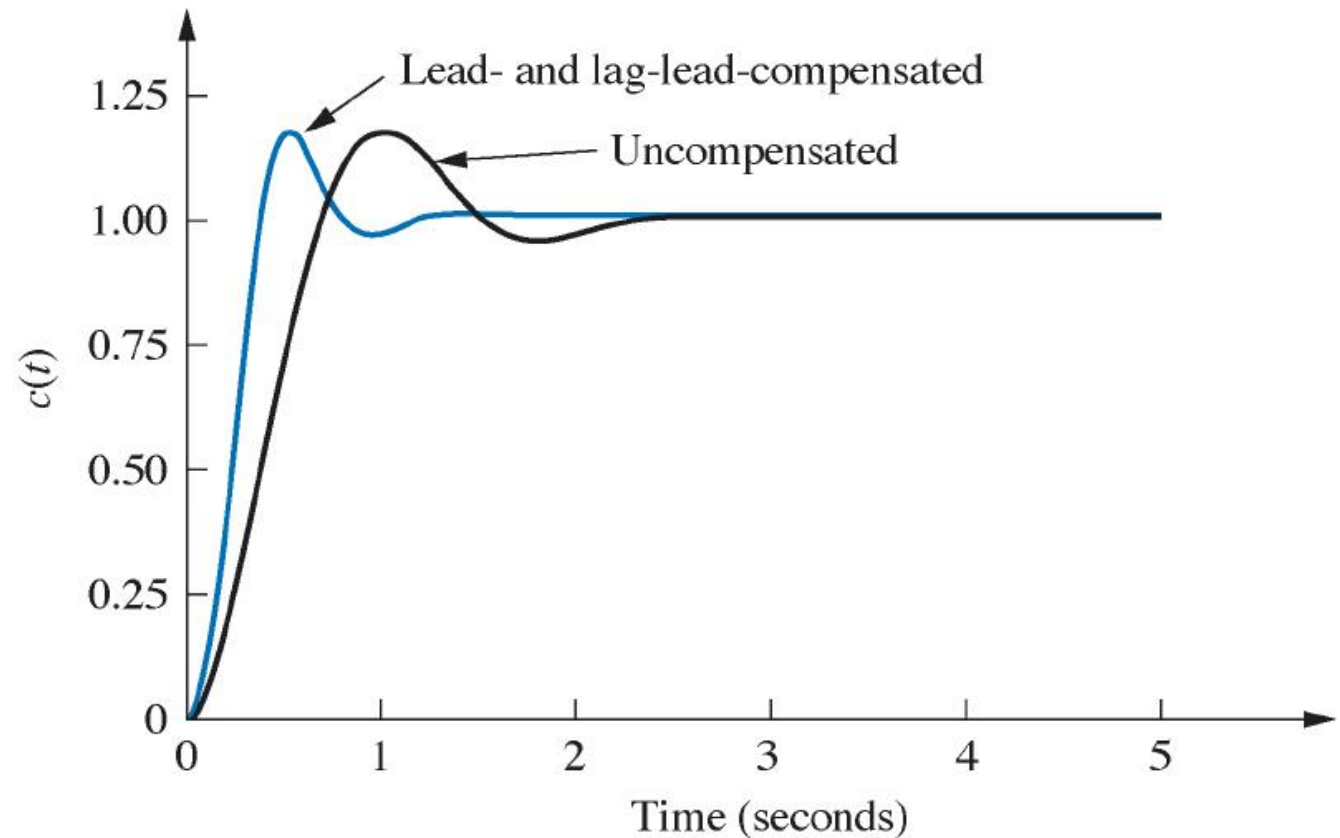
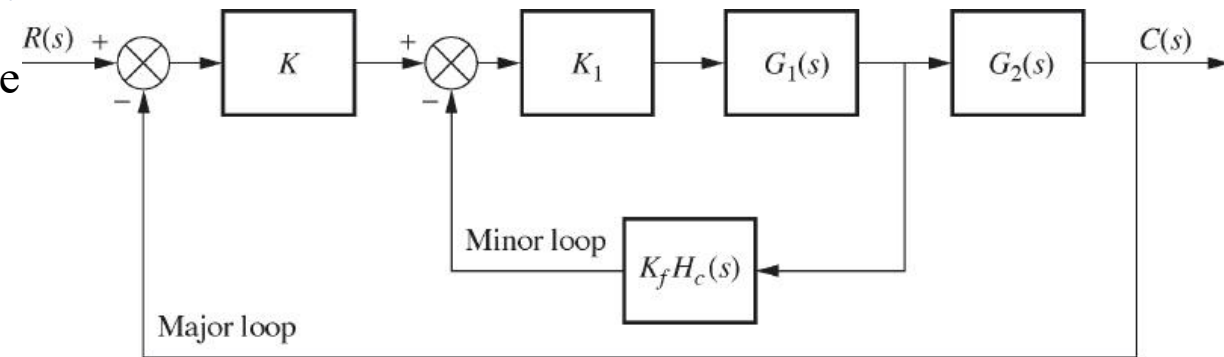


Fig (b) Improvement in step response for lag-lead-compensated system

Feedback Compensation

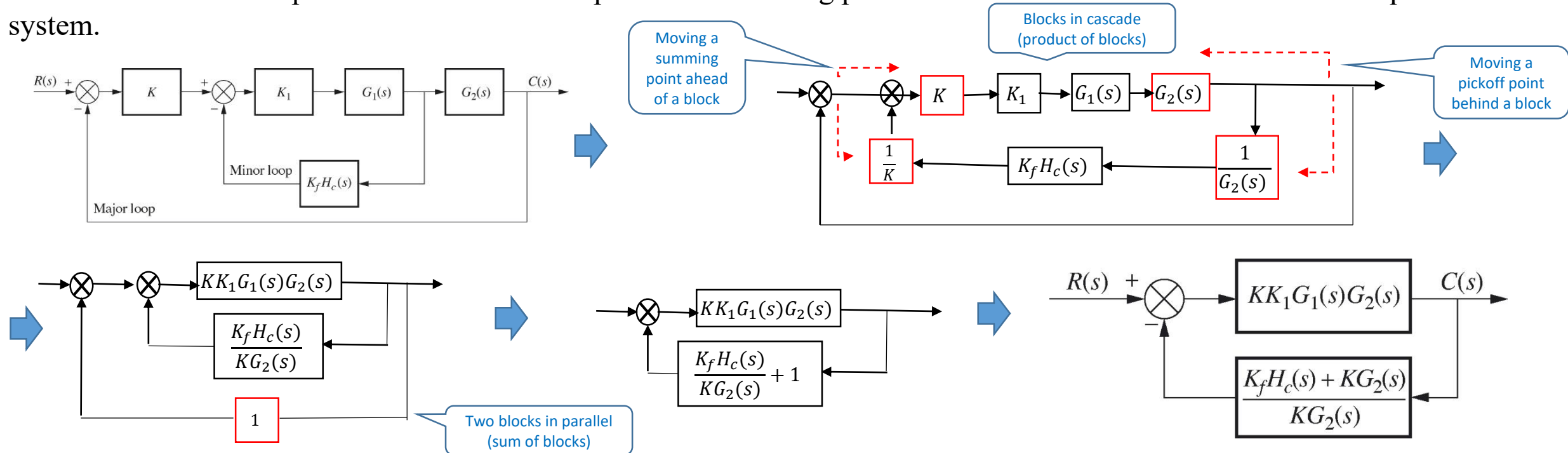
Compensator $H_c(s)$ is used at the minor feedback to reshape the root-locus and improve transient response and steady-state response independently ($G_2(s)$ can be unity).

- Can be more complicated than cascade.
- Can provide faster response.
- Can be used in cases where noise is a concern if we use cascade compensators.
- May not require additional gain.



The design of feedback compensation consists of finding the gains, such as K , K_1 and K_f .

Similar to cascade compensation. Consider compensation as adding poles and zeros to feedback section for the equivalent system.



Example 7₁

Given the system of Figure (a), design rate feedback compensation, as shown in Figure (b), to reduce the settling time by a factor of 4 while continuing to operate the system with 20% overshoot.

SOLUTION

- First design a PD compensator.
- For the uncompensated system, Search along the 20% overshoot line ($\zeta = 0.456$)

The angle of the 20% overshoot line



$$180^\circ - \arccos(\zeta) = 117.13^\circ$$

the dominant poles



$$-1.809 \pm j 3.531 \text{ (see fig (e))}$$

- The settling time is 2.21 seconds and must be reduced by a factor of 4 to 0.55 second.

- Next determine the location of the dominant poles for the compensated system.

- To achieve a fourfold decrease in the settling time, the real part of the desired pole must be increased by a factor of 4.

Real part of Compensated pole

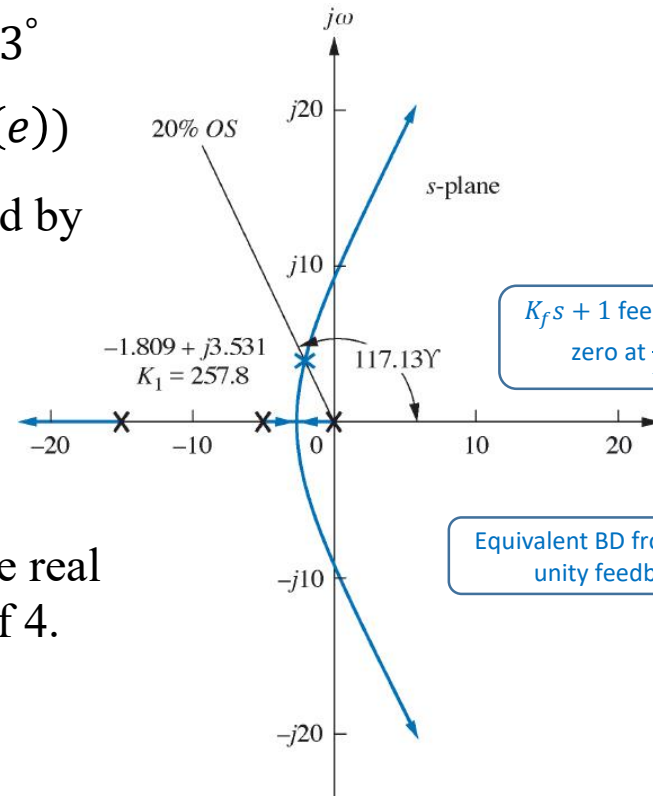


$$4(-1.809) = -7.236$$

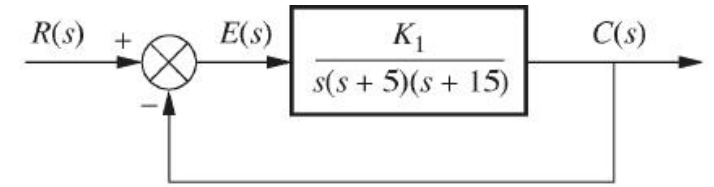
Imaginary part of Compensated pole



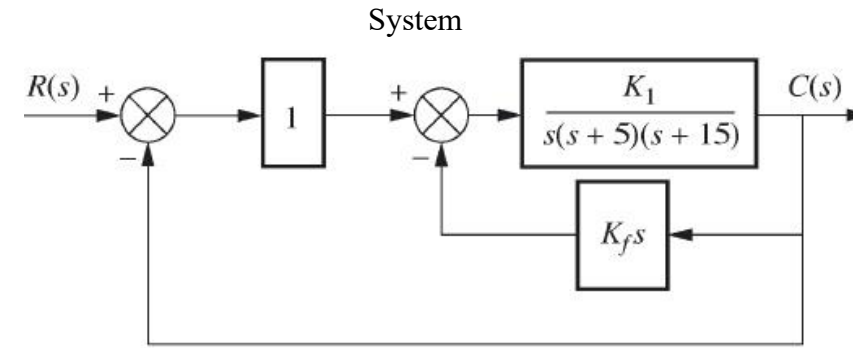
$$w_d = -7.236 \tan(117.13^\circ) = 14.12$$



(e) Root locus for uncompensated system

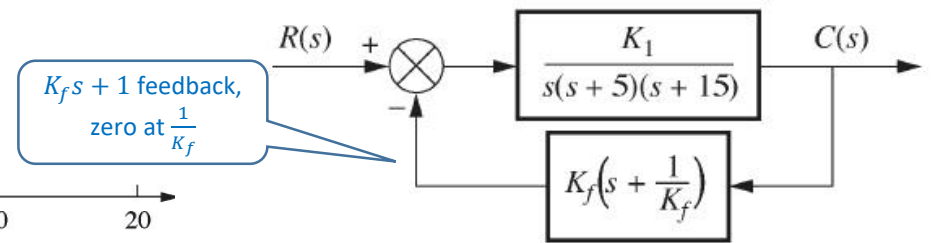


(a)



(b)

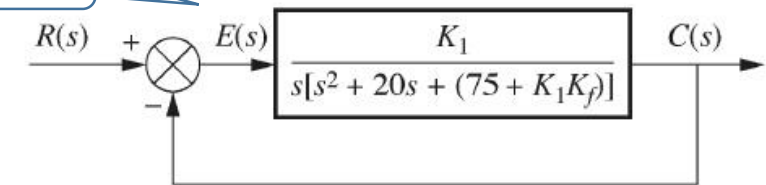
system with rate feedback compensation



(c)

equivalent compensated system;

Equivalent BD from fig (b)
unity feedback



(d)

equivalent compensated system showing unity feedback

Example 7₂

Compensated dominant pole position



$$p_c = -7.236 \pm j 14.12$$

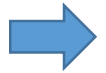
- Sum of the angles from the uncompensated system's poles (add zero to yields 180°)

compensator zero contribution

$$\theta = -277.33^\circ \quad \text{blue arrow} \quad \theta_z - 277.33^\circ = -180^\circ \rightarrow \theta_z = +97.33^\circ$$

- Using the geometry shown in Figure (a)

Compensator's zero location



$$\frac{14.12}{7.236 - z_c} = \tan(180^\circ - 97.33^\circ) \quad \text{blue arrow} \quad z_c = 5.42$$

- The root locus for the equivalent compensated system (fig (c) previous slide) is shown in Figure (b)

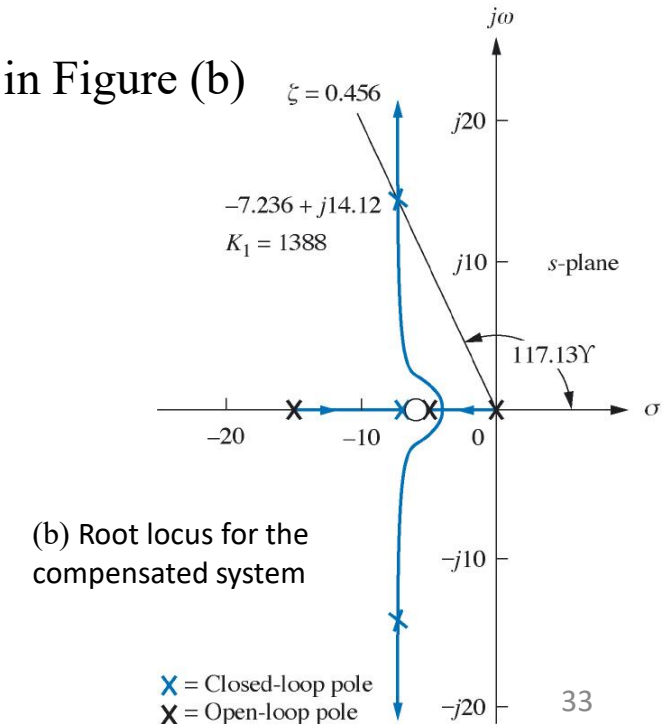
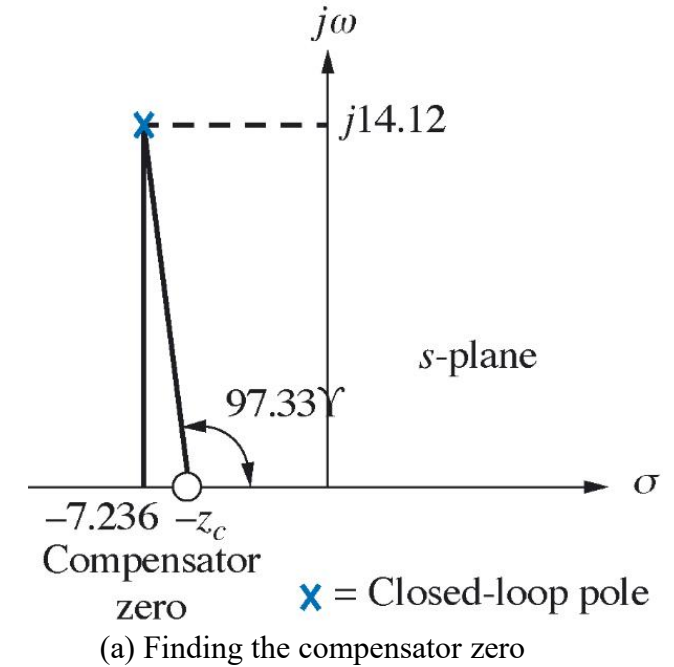
The gain at the design point, $K_1 = 1388$

Since K_f is the reciprocal of the compensator zero,

$$\text{blue arrow} \quad z_c = \frac{1}{K_f} \quad \text{blue arrow} \quad K_f = \frac{1}{z_c} = \frac{1}{5.42} = 0.185 \quad \text{blue arrow} \quad K_1 K_f = 256.7$$

- steady-state error characteristic (fig (d) slide 32)

$$K_v = \lim_{s \rightarrow 0} sG(s) = \frac{K_1}{75 + K_1 K_f} = 4.18$$

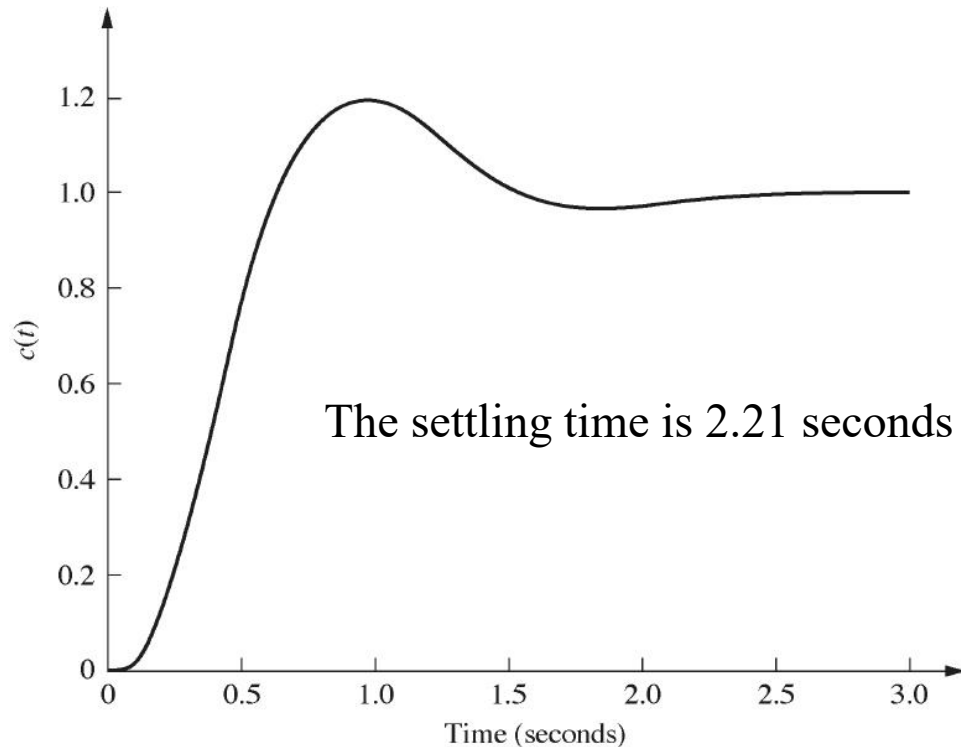


Example7₃

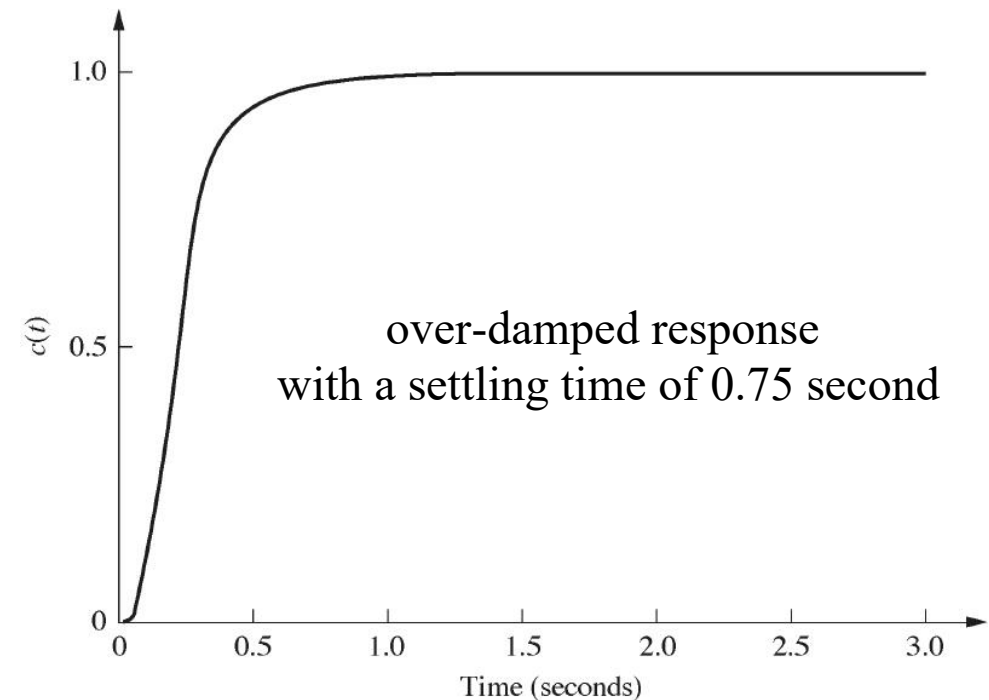
- The closed-loop transfer function is (fig (d) slide 32)

$$T(s) = \frac{G(s)}{1 + G(s)H(s)} = \frac{K_1}{s^3 + 20s^2 + (75 + K_1K_f)s + K_1}$$

- The results of the simulation are shown in Figure (a) and (b)



(a) Step response for uncompensated system



(b) Step response for the compensated system

Physical Realization of Compensation

Active-Circuit Realization

- $Z_1(s)$ and $Z_2(s)$ are used as a building block to implement the compensators and controllers, such as PID controllers.
- The transfer function of an inverting operational amplifier

$$\frac{V_o(s)}{V_i(s)} = -\frac{Z_2(s)}{Z_1(s)}$$

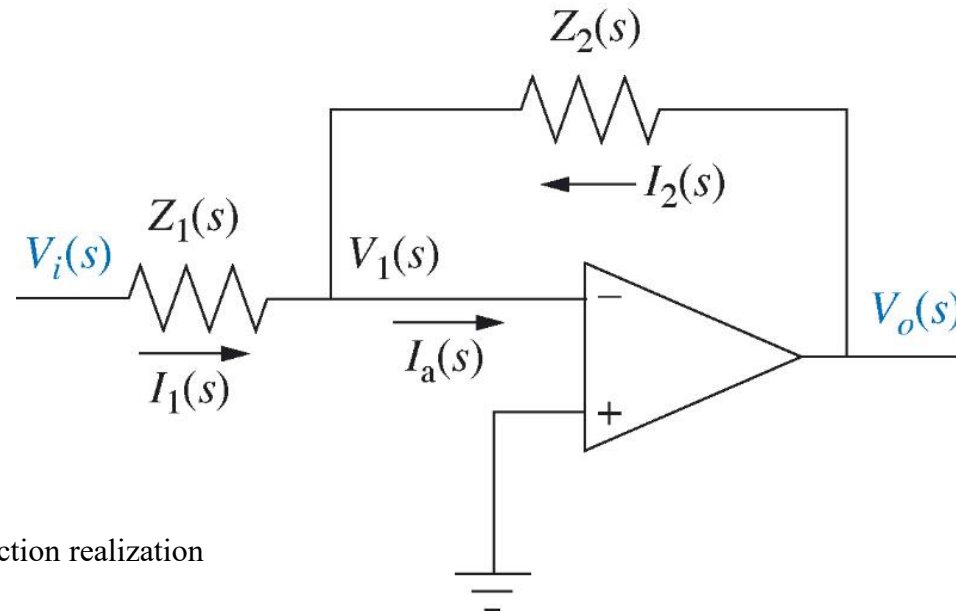


Fig (a) Operational amplifier for transfer function realization

- Table1 summarizes the realization of PI, PD, and PID controllers as well as lag, lead, and lag-lead compensators using Operational amplifiers.
- Fig (a) : lag-lead compensator can be formed by cascading the lag compensator with the lead compensator.

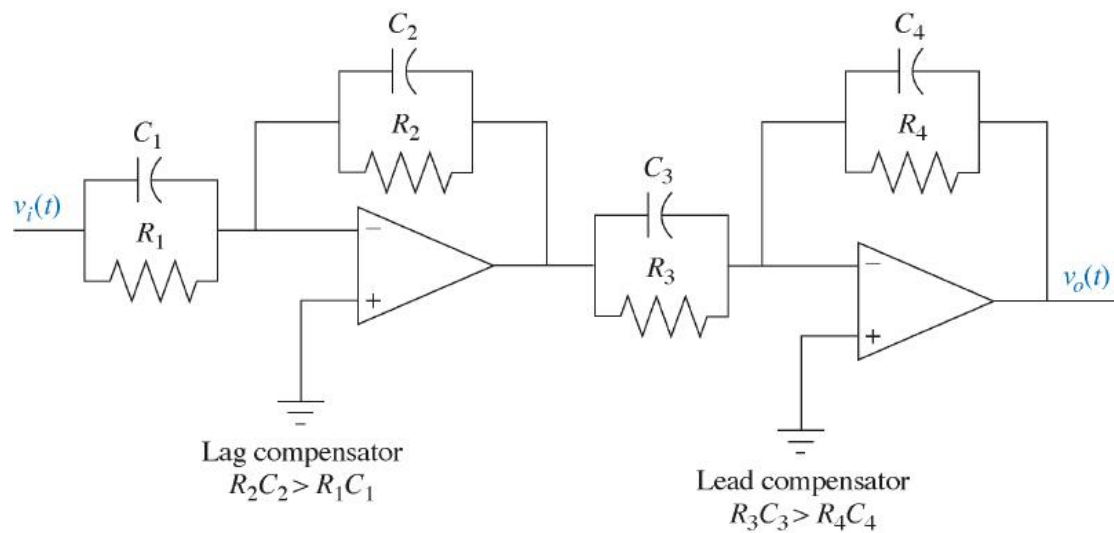


Fig (a) Lag-lead compensator implemented with operational amplifiers

Function	$Z_1(s)$	$Z_2(s)$	$G_c(s) = -\frac{Z_2(s)}{Z_1(s)}$
Gain			$-\frac{R_2}{R_1}$
Integration			$-\frac{1}{RCs}$
Differentiation			$-RCs$
PI controller			$-\frac{R_2}{R_1} \left(s + \frac{1}{R_2C} \right)$
PD controller			$-R_2C \left(s + \frac{1}{R_1C} \right)$
PID controller			$-\left[\left(\frac{R_2}{R_1} + \frac{C_1}{C_2} \right) + R_2C_1s + \frac{R_1C_2}{s} \right]$
Lag compensation			$-\frac{C_1}{C_2} \left(s + \frac{1}{R_1C_1} \right) \left(s + \frac{1}{R_2C_2} \right)$ where $R_2C_2 > R_1C_1$
Lead compensation			$-\frac{C_1}{C_2} \left(s + \frac{1}{R_1C_1} \right) \left(s + \frac{1}{R_2C_2} \right)$ where $R_1C_1 > R_2C_2$

Example8

Implement the PID controller of Example 5

SOLUTION

- The transfer function of the PID controller is $G_c(s) = \frac{4.6(s + 55.92)(s + 0.5)}{s}$
- which can be put in the form $G_c(s) = s + 56.42 + \frac{27.96}{s}$
- Comparing the PID controller in Table 1 with this equation we obtain the following three relationships:

$$\frac{R_2}{R_1} + \frac{C_1}{C_2} = 56.42 \quad R_2 C_1 = 1 \quad \frac{1}{R_1 C_2} = 27.96$$

- Rhmbd sgddq ` qd ent qt nj mv nr ` ne sgdd dpt ` slmr
we arbitrarily select a practical value:

$$C_2 = 0.1 \mu F \quad \longrightarrow \quad R_1 = 357.65 k\Omega, R_2 = 178.891 k\Omega \text{ and } C_1 = 5.59 \mu F$$

- The complete circuit is shown in Figure (a) where the circuit element values have been rounded off.

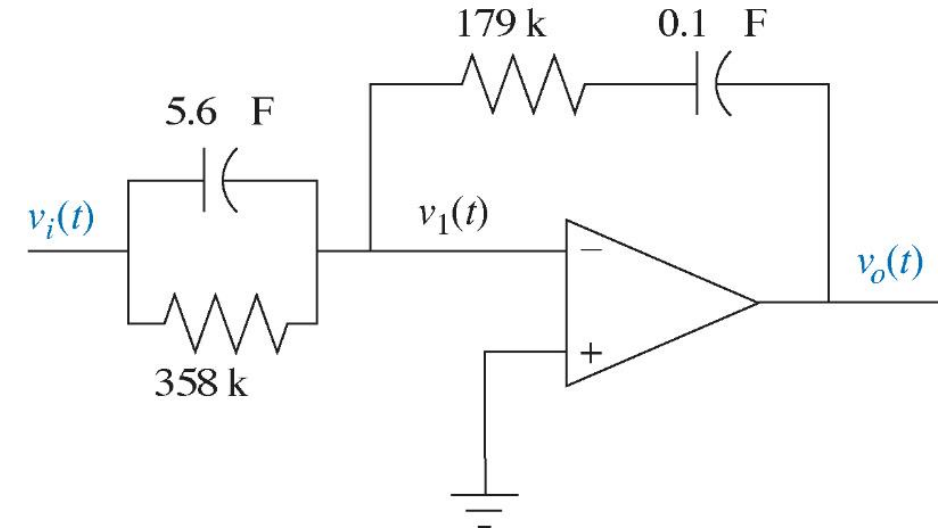
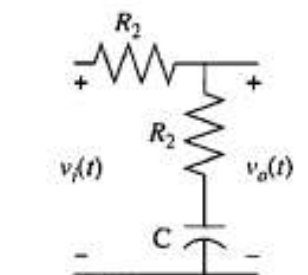
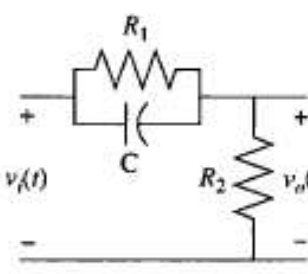
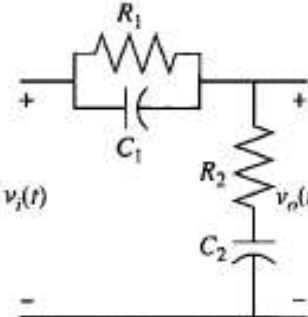


Fig (a) PID controller

Passive-Circuit Realization

- Lag, lead, and lag-lead compensators can also be implemented with passive networks (Table 2) .

Function	Network	Transfer function, $\frac{V_o(s)}{V_i(s)}$
Lag compensation		$\frac{R_2}{R_1 + R_2} \frac{s + \frac{1}{R_2 C}}{s + \frac{1}{(R_1 + R_2)C}}$
Lead compensation		$\frac{s + \frac{1}{R_1 C}}{s + \frac{1}{R_1 C} + \frac{1}{R_2 C}}$
Lag-lead compensation		$\frac{\left(s + \frac{1}{R_1 C_1}\right) \left(s + \frac{1}{R_2 C_2}\right)}{s^2 + \left(\frac{1}{R_1 C_1} + \frac{1}{R_2 C_2} + \frac{1}{R_2 C_1}\right)s + \frac{1}{R_1 R_2 C_1 C_2}}$

Example9

Realize the lead compensator designed in Example 4 (Compensator b zero at -4).

SOLUTION

- The transfer function of the lead compensator is $G_c(s) = \frac{s + 4}{s + 20.09}$
- Comparing the transfer function of a lead network shown in Table 2 with The equation, we obtain the following two relationships:

$$\frac{1}{R_1 C} = 4 \quad \text{and} \quad \frac{1}{R_1 C} + \frac{1}{R_2 C} = 20.09$$

- Since there are three network elements and two equations, we may select one of the element values arbitrarily

$$C = 1 \mu F \quad \Rightarrow \quad R_1 = 250 \text{ k}\Omega \quad \text{and} \quad R_2 = 62.2 \text{ k}\Omega$$